

# AVIATION WEEK

A MCGRAW-HILL PUBLICATION

SEPT. 19, 1949

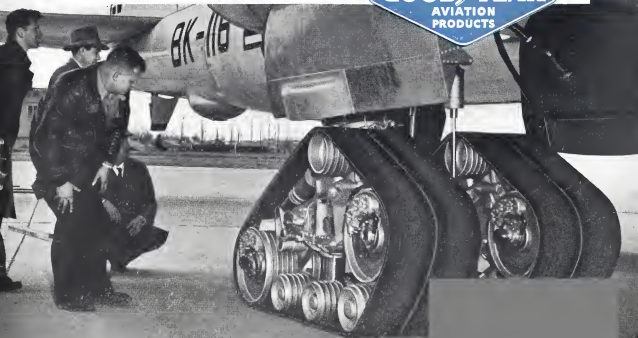
## Now a rubber runway for the B-50

TO solve the problem of handling the giant B-50 Boeing Superfortress on unimproved fields, Goodyear, in cooperation with Boeing engineers, has developed a radically new retractable landing gear—the world's first track-tread landing gear for a heavy four-engine bomber. This unique gear multiplies the practical operating field of giant planes because it increases the footprint area three times that of the conventional landing gear of the B-50. Interesting design features include the use of

endless rubber belts, reinforced with steel cable, running over a series of floating bogies controlled by a battery of four Goodyear Single Disc Brakes on each main track—the entire assembly furnished by Goodyear! This revolutionary landing gear for giant bombers is another of the many Goodyear-built products which have spurred aviation progress for the past forty years.

Goodyear, Aviation Products Division  
Akron 16, Ohio or Los Angeles 54, Calif.

MORE AIRCRAFT LAND ON GOODYEAR TIRES, TUBES,  
WHEELS AND BRAKES THAN ON ANY OTHER KIND





Safe, accurate  
and dependable  
controls  
promote the future  
of aviation.



**CREATIVE ENGINEERING**

Makers of the Famous Multi  
Electronic Autopilot, Fuel Meter,  
and Turbo Supercharger Con-  
trols, Standard on Many Types  
of AAF Aircraft



MINNEAPOLIS  
**Honeywell**  
AERONAUTICAL CONTROLS

*Nothing Rolls Like a Ball...*



*Reflecting the experience,  
resources and skill of the  
world's largest producer  
of fine ball bearings.*

**NEW DEPARTURE  
BALL BEARINGS**

NEW DEPARTURE - Division of FEDERAL MOTORS - ROSSIER, CONNORLEY



# PLANE FAX

## THRIFT TIPS

How to reduce flying expenses

A page of service tips for private flyers and fixed-base operators

## SAVE FUEL

### Use Improved Chevron Aviation Gasoline (80/87)

You'll enjoy fuel economy and maximum power with Chevron Aviation Gasoline (80/87)—that brand new fuel with the controlled takeoff and knock saving. This fuel virtually eliminates the possibilities of take-off detonation... means even combustion without waste... smoother, quieter, safer take-offs than you've ever known... less wear on vital engine parts. And at an extra cost.

Extra savings go to pilots of larger aircraft whose engines also require 91/96 fuel. Improved Chevron Aviation Gasoline (90/97) already has been approved by Pratt & Whitney Aircraft for use in their Whop-Jet engines. Now you can cut expenses two ways—lower fuel costs and lower maintenance costs. That's because of knock-free operation and reduced lead content which means less, cleaner operation.



Cut overall costs and engine wear—use "RPM"

Many pilots double the time between overhauls and cut down expensive repair bills by using RPM Aviation Oil. They've found that ordinary oils can wear from hot upper cylinder walls, and hot pistons grind and scrape. But RPM is specially compounded to cling to hot spots most able to burn them, keeping a protective oil coating over all engine parts.



CHEVRON  
AVIATION  
GASOLINE

RPM  
AVIATION OIL

Standard Oil Company of California

"It takes better care of your plane"

## NEWS DIGEST

### DOMESTIC

ADAPAA merger agreement, due to expire last week, was extended until Mar. 13, 1948, with provision for the three-month extension. A change in the past practice that TAA will pay \$77,418,000 cash for ADA assets and assume ADA's liabilities as of closing date. Original agreement provided for stock exchange rather than purchase by cash.

Eastern Air Lines has asked CAA permission to operate 4-seat per side coach service from New York to Miami and New Orleans, starting Sept. 15. CAA probably would use 36-passenger DC-4s.

Swedish Saab Scania, 34-seat piston-type transport, opened U. S. test for climbing tests at East Hartford, Conn. After stop at New York, it was delayed for seven months to Washington, Alaska. Chicago (on Sept. 13, 194), Minneapolis (21), Kansas City (21), St. Louis (21), Dallas (21), Houston (21, 23) and Oct. 1. On Oct. 4, it is scheduled for Atlanta, showing, Miami on 5, 6 and 7, then returning to New York via East Hartford for light tests in the South America. CAA is powered by P&W R-2600s, but previous versions will have R-2800s. Conquest Aircraft Corp. may be interested in producing Scania here.

Northwest Airlines plans to start St. Petersburg service to Miami on Nov. 1. United Air Lines' August passenger traffic was up 10 percent over same month in '45, and slightly above July '49.

Lockheed Constellation became the world's first craft certified for international operation in compliance with International Civil Aviation Organization's new standards, which became effective Sept. 1.

Navajo Aircraft Division of United Aircraft Corp. marked 10th anniversary of VS-100's first flight, the month. More than 600 of the company's helicopter processors to the craft have flown in military and commercial service, piling up in excess of \$6,000 in.

George W. Elcock, founder and president of Dayton Aircraft Products, Dayton, Ohio, died at Portland, Ore.

New CAA August Airlines Committee elected A. J. Curry, Miami, chairman, and Walter E. Betts, Washington, vice-chairman, at the opening of its third year in Washington. Next meeting will be in Los Angeles in January. Committee heads discussions on airport needs from representatives of

airlines, private pilots, Post Office Department, and state aviation officials and Administrator DeLoe W. Renshaw.

### FINANCIAL

National Airlines reported \$18,963 profit for fiscal period ending June 30, against loss of \$3,309,000 for fiscal '48. Fiscal '49 losses were a high as \$1.5 million, and profit was achieved after absorbing costs of DC-6 grounding, also strike of pilots and ground employees.

Northwest Airlines is committed to start raising its \$1,750,000 RFC loan, with \$12,000 monthly payments, on Nov. 16. Under loan agreement, NEA may not deliver dividends, consolidate, merge, or make purchase or construction commitments of over \$100,000 without RFC approval.

All American Airways reported \$755,750 net loss for fiscal year ending June 30, compared with \$244,140 deficit for fiscal '48.

Pratt & Whitney Co. shows net loss of \$15,627 on sales of \$7,945,603, a record report for their ending June 30, compared with net income of \$180,528 on sales of \$5,585,415 in '45. Loss or profit reduction in inventory of about \$600,000 charged to year's operations and approaching write-downs to net worth included in net loss in inventory of some slow-moving items and other high-cost work being discontinued. Total of \$214,404 was spent on machinery and other capital improvements. \$307,100 for new Los Angeles plant.

### INTERNATIONAL

Czechoslovak Airlines will receive further assignment of eight Douglas 124. Two of these craft are in regular service in Prague from time.

Italian Fleet Captain John Emswille and Captain Ransford, backed into Bombardier at Laguna Airfield, Arizona, when an air traffic control short their attempted exit west loop over the Atlantic from London.

Quebec Airways DC-3 craft on which traffic was St. Louis, Que. is held at 27th Street, CAA is subsidiary of Canadian Pacific Air Lines.

Milwaukee Air Transport Service is serving 30 scheduled G-41s to Cook Air Force. Flights were passed from military service in U. S., will be used for under Guard Air program.

Polish Airlines LOT has purchased 26 planes from Russia. They are reported to be "last modern Soviet planes, equipped with radio."

## Edo NEWS NOTES

32/1000s, plus 10 miles, a record and close quarters for most previous aircraft short range work. But such advances are chiefly paid in the Edo machine who are doing research for the Army people under modern conditions as never before. Many parts of these complex assemblies may be automated within year or more 32/1000s of an inch.



Edo has had 20 years of experience in manufacturing machine tools, which must be made right as well as accurately doing, and from the machine in operation, an example of how well qualified Edo is in handling such work, and how accurate, and in fact.



Used on the Edo machine which does the B-50 bomber, the machine has done 100 parts in three months, made 100 parts each and are made a yard long. They must meet size and weight, balance and finish of size more than 1000s of an inch.

Currently hundreds of the process are being manufactured by Edo for the Army, just one of a wide variety of machine fabrication jobs handled by Edo.

Look to EDO for:

- ELECTRONIC DEVELOPMENT AND MANUFACTURING
- SEAPLANE FLOATS
- FLEXIBLE ALUMINUM FABRICATION
- DESIGN DEVELOPMENT AND ENGINEERING

Edo

Sole U.S. Distributors  
1000 N. 10th St., Suite 100, Phoenix, Arizona



If it's IBM...  
it is electric



As natural as the application of electric power to communications or mass production is the application of electricity to typing.

IBM, pioneer in this development, has engineered the IBM Electric Typewriter to bring economy to the modern office through the advantages of speed, uniform quality, and ease of operation.

A light touch operates all keys on the

IBM Electric Typewriter—including carriage return, tabulator, backspace, and shift. Uniformity of appearance—regardless of the typist's touch—is insured by the built-in depression control. The easily-adjusted multiple copy control provides for one or many legible carbons.

The IBM representative nearest you will demonstrate the many new features of the IBM Electric Typewriter.

**IBM**

INTERNATIONAL BUSINESS MACHINES CORPORATION

World Headquarters Building, 390 Madison Avenue, New York 22, New York



## Newest British Aircraft Revealed at Show

### Turbine-powered military and transport planes give spectacular flight displays.

By Robert Ross

**LONDON**—The latest in British transport and military aircraft caused all the very best aviation travelers at Farnborough last week in a five-day display to speculate a British sales campaign aimed at exporting \$112 million in planes and engines by 1953.

Opening at the 10th annual Society of British Aircraft Constructors exhibition was kickstarted by appeals from British government leaders to push aircraft export sales to a new peak, renewed confidence of British aircraft builders in their commercial products, and a strong British press machine warning readers that "British aircraft again lead the world."

"Best in World"—The show sample evidence that an dependence on U. S. aircraft is a phase that is now passing. There are places on our map that cannot be better serviced elsewhere in the world and there are other planes coming far ahead that society prefers them from being shown this year.

Since British manufacturers took a more sober view, but generally felt they have now emerged from their postwar slump in civil aviation development and are in a position to give American transport builders stiff competition in international markets. Visitors observers of the SBAC show agreed that the 1949 display far surpassed any other since

the war particularly in the number and quality of civil transports in the flying stage.

**Foreign Delegations**—The SBAC attracted several thousand foreign military and civil aviation experts from 54 countries, representatives of 46 airlines, and roughly a quarter million Britons on the two public days. Among the overseas contingent viewing the show were Harold Harro, president, and Larry Tate, operations director, of American Overseas Airlines; Rear Adm. Alfred M. Pele, chief of the Navy Bureau of Aeronautics; Don Briggs, director of the Navy's aircraft design research division; Maj. Gen. Carl Brundage, chief of U. S. Air Force requirements division; Maj. Gen. Donald Pitt, USAF director of research and development; Maj. Gen. Louis Johnson, commander of the Third Air Division; George B. Woods, special assistant to USAF Undersecretary Arthur Brown; and a large group of U. S. military and civil air officials and Western Union defense staff members.

Far most American observers the two attractions were the gas turbine-powered experimental aircraft and the mighty Rolls-Royce Avon turbojet engine now officially rated 17,500 lb static thrust. Unraveling of the solid flow Avon, which is now being in the Gloster two-gig bomber and a Gloster Meteor fighter indicates that the British are still maintaining a major role in jet engine manufacturing in the world field. Most powerful U. S. jet engine in a compar-

able state of operational use is the General Electric J-47 rated at 12,000 lb, but now actually delivering 16,000 lb static thrust.

**Coastal Skuas**—Brightest star in the British transport galaxy, is of course, the de Havilland 16-passenger Comet powered by four DH Comet (1900 lb thrust) turbojets. Considerably smaller than the Douglas DC-6 and the Boeing Page Fourteen series, the Comet is one of those planes that cannot be fully appreciated until it is seen flying.

Although it was being mostly empty at Farnborough (except for fuel and test maneuvers) its flight performance in the hands of DH chief test pilot John Cunningham gave observers a good appreciation of its flight characteristics.

De Havilland a full aircraft section on details of the Comet's construction and performance, but the first 15 hours of test flying have seen the jet airliner impose its designed cruising speed of 500 mph and its anticipated cruising altitude of 40,000 ft. Comet has reached Mach 0.8. Observers were impressed with its short turnoff on incompressible less than that of an F-56 and the moderate approach and landing speeds.

It is about from the long, low wing combining 30 deg of sweepback with high taper that the Comet has a wing loading probably not so high as some American transports now in service. Approach and landing are aided by a tremendous flap area extending outward from the wing to the fuselage. The flaps include a special arrangement whereby the outer side of the jet engine nacelles are split laterally and the bottom half lower to become part of the flap area.



COMET. Shows that in Britain's turbulent waters, even the stars are being repositioned.

Cunningham, who has done all of the Coast flying to date, credits the Coast with the outstanding main runway at Fairbairn as a factor in nearly perfect landings that gave a *disappointingly* light exposure of the landing aspect in the Cunningham documented rate of climb, maneuverability and speed with a low level buzz across the field at 250 mph.

► **Cowled Appearance**—Combination of high input rates and incoherence give the Cassini an unusually graceful appearance in the air traffic load in workload transports. The posted, downmost note gives it an atmosphere of speed like that achieved by a similar analysis in the Lockheed Contribution.

The Ghost is not a particularly radical-looking glass. De Haverland officials indicated that they modified conventional design plans to retain more conventional features considered to maximize prospective passenger. Of special interest is the manner in which the 55-in. diameter Ghost tapers to a base barely based in a thin wing. There is little nacelle protrusion. In a head-on flight view it is hard to spot the Ghost's existence.

Drug profile is extremely close. Dr. Hayland is already well along on production of the initial order for 10 Concorde-tuos for the Ministry of Supply and 14 for British Overseas Airways Corp.

Tentative plans of BOAC indicate the Comets will get their initial sailing proving on the Empire routes to Africa and Australia rather than on the blue ribbon North Atlantic run.

- **Turbosens Plants-Nut** is the Canist. Design interest centered on the latest collection of turbosens assessed within our deployed. They include:
  - Miles Marathon-20 connector 354

sph. linerliner powered by two 1400 hp. ArmstrongSulley Maske two sets

- **Viking Vascourt**—12 passenger, 275 mph, fast powered by four Rolls-Royce 1500-hp Dart turboprops
- **Academy-Whitcomb Apollo**—11-passenger 305 mph, fast, powered by four 1600-hp Pratt & Whitney turboprops

\*Händler: Page Motors V-60 power get. 146-cu-in. fuel powered by four 2400-hp. Detroit Diesel technology.

The 35,000th Hermes V is the largest turbo-prop aircraft ever flying. It made its first flight only a few weeks ago and its first public appearance at Farnborough. Comparison of the Hermes V with its predecessor counterpart the Hermes IV should provide interesting data on the relative operational merits of the two types of powerplants by analysis. The Hermes V has a thinner and lighter wing than the Hermes IV which was the lowest military type wing developed for the Hawker Fury Hermes.

The turboprop Hercules IV goes 3000 ft more than the V and cruises nearly 2500 ft more per hour. It weighs at 62,000 lb, faster than the piston-powered Hercules, climbs nearly 1000 ft, faster and has a service ceiling 1000 ft higher. The Hercules wing is slightly smaller than the DC-8.

► **Apollon** in Vietnam—Pescadore operations for the immediate foreign air-line market are the Vietnam and Apollon. Vietnam is giving sole production on a 40 passenger larger version (Model 700) of the Vietnam with an order for 50 transports—20 for British European Airways and 12 for BOAC. Initial Vietnam service is expected to be an B747. Continental services and on BOAC's West India routes. Vietnam also has plans for a 140 passenger version of the

**Vacuum 700.** Craning speed on the production Vacuum will be boosted from 275 mph. for the prototype to 325 mph. with a payload boost of nearly 5000 lb. Gross will be increased from 48,400 lb. on the prototype to 45,000 lb. on the Model 700.

The Vauxhall would be competing in the Covent-Lineer class for the fastest market and is expected to sell for about \$190,000 in contrast to the present Covent-Lineer price of about \$250,000. Australian National Airlines is extremely interested in the Vauxhall and is expected to close a deal with Vickers for 10 Vauxhall Model 700s shortly.

The Apollo impressed foreign observers in perhaps a more advanced manner than the Vought but development delays of the Marbha engine seemed to enter into that rating for nearly a year. The Apollo has a heavy stressed and stout air which should produce good directional stability. British pilots who have flown the Apollo confirm this characteristic, particularly when flying with two engines inboard on the same side.

Like the Viscount the Apollo is always passengers with its substantial British and low noise level. Belgium SARINA and the Argentine FAMA are lines have been particularly interested in buying the Apollo and preliminary negotiations are under way for initial sales.

With a gross of 48,000 lb. for Apollo office payloads of 7900 lb. and comes at 40¢ each over a range of 1000 miles. This compares with an 1800-cc. move for the Vauxhall 700.

**Manuka**—The Manuka offers an either interesting contrast between a patch and battery-powered version of the same engine. Substituting two Manuka batteries for the four Gages Quam parts means increased total horsepower by 600 reduced gear weight by 220 lb and increased engine speed by 45 mph. There are to set up five orders for the Manuka and two Manuka either four British or foreign orders.

As the turboprop liners awaited take-off they sounded more like robust steam engines than aircraft. All of the large turboprop airliners have demonstrated robust and dumb characters: start with two of their four propellers feathered on the same side. The Manitoweg flew and landed on a single engine.

Britain's latest transport, the Bristol Brabazon 1—as later versions also slated to use turbofans—flew over the show area at 180 mph and 500 ft. altitude, piloted by A. J. Rose.

UNBENTENK significant improvements not obvious, but lower prediction errors. Model 7B, will be even better

song and somewhat conventional configurations on the ground. But in the air, the combination of test pilot R. F. Beumont and the 15,000-lb. thrust from the two dual A4s made the Canberra behave in spectacular fashion.

In speed trials from 730 to less than 100 mph, we ably demonstrated by instrument what followed by high speed pilots on the deck with an approach using full flaps and gear down and hand-by doors open that slowed the Crusers to less than 100 mph. At this speed he sucked the big bomber violently with a desire to show the full control available as it approached shell-on-shell.

Bombardier whipped the bomber (it aimed to carry a 10,000-lb. bomb load) around on the deck like a fighter, flying it through a series of slow rolls, high-speed turns and considerable rates of climb. The Corsair was originally designed for rate bombing at around 30,000 ft but Bombardier's demonstration convinced many testiners the new bomber may prove to be another Mustang in its suitability at everything from low-level attack through high-

► **Light Bomber**—According to USAF standards the Canberra is classified as a light bomber, a field in which there have been no outstanding American postwar planes. U. S. Ground Army high command has been soliciting USAF to heavy pressure to develop a low-level, very light bomber that could

The Canbera has a span of 66 ft., length of 65 ft., 6 in. Tail stands 15 ft., 2 in. above the ground. Wings have a

English Electric a building 12 for Royal Air Force stores depot.

**Delta Wing**—Most modern British design on display was the Aero Model 757, a Delta wing research plane. The Aero Delta is roughly comparable in general appearance to the Northrop X-4 and Cancon XP-92A. Its Delta wing is limited slightly up on the fuselage and has a thick slot inboard as a flap up. The Aero Delta has a long dorsal fin but small saucer and vertical fins. An intake for the 3500 lb thrust Derwent 3 turbojet are located on top of the fuselage behind the cockpit. This intake is bled off by the dorsal fin.

Outboard sections of the wing built with the inboard sections lowering on three supports at Aft. Wing span is 34 ft and foreleg is 30 ft, 6 in. long. The Aero Delta had had only two hours flying time when it appeared at Farnborough. It is being used in a research plane for an Aero Jet Delta wing booster project. The Aero jet booster is scheduled to be considerably bigger than the Cariburn and will fall into the USAF classification of a medium bomber.

Among the other outstanding military aircraft on display were

• **De Havilland Vampire (DH-112)**—The latest version of the Vampire featuring a 50 percent increase in power and aerodynamic modifications for flight at high Mach numbers. The power plant is a DH Ghost turbojet (1400 lb thrust). A new high-speed serial has been substituted for the normal Vampire wing. Jetisonable wing tip tanks have been added for the

first test in the Vroom series to a coastal range. The Vroom is on the 650 mph plus class and maintains its operational efficiency into the 45,000-50,000 ft altitude range.

•The Hawford Night Fighter (DHL

1130-This is the first British jet night fighter and is based on a standard Vampire design. Modifications include a longer cockpit to accommodate a radar operator, an elongated engine nose to house subsonic radar gear, more wheel strut moved back slightly to give the radar nose more ground clearance, and additional fuel capacity. The night fighter, which was entered into production this year, is powered by the 1920-hp. short 1949 Cobalt engine.

World War II. Wyvern—This is the first Egger powered solely by a tailplane. It mounts an Armstrong-Whitely 400-hp. Pylons tailplane in the nose, turning counterbalancing inboarded propellers and pushing air thrust through tail pipes just above the wing on both sides of the fuselage. The tailplane is a 100-hp. Egger 100-hp. The 100-hp. Egger is an operational aircraft and has just passed its 175-hour type certification for both civil and military use. The model-bird Wyvern is being built for career service with the Royal Navy and is scheduled to begin carrier trials in October. Present plans call for registration of the carrier air group with the Wyvern.

• **Glacier Master Mask VIII**—This is the latest version of the Mettore series and features a re-designed tail aimed at better stability at high Mach numbers, an elongated nose adding a fuselage fuel tank, and three removable fuel tanks two on the wing tips and the third fitted under the fuselage similar to the tank on the McDonnell Phantom.

(Continued on p. 16)

## Society of British Aircraft Constructors Shows . . .



Avro 707



Avro 707



Avro 707



DH113



DH112 Venom

## . . . Its Bids for World Air Leadership



Apollo



Conquest



Endeavour 1





WYVERN Fast biplane fighter has most powerful biplane engine

## Farnborough Show

(Continued from p. 15)

► **Fighter Trends**—The British fighters now continue broadly toward the development of the Vampire and Meteor designs, through aerodynamic refinements and increased power to produce high speeds and high altitude performance required of modern jet fighters. As yet unannounced publicly, but definitely on the way, is a British version of the rocket-powered interceptor type fighter. Decisions to concentrate basic RAF production on the two standard types meant that nothing in the Vulcan swept-wing Type 516 or the Hunter (swept-wing P. 1052) is largely a matter of cost—it being considerably cheaper to modify the Vampire and Meteor rather than tool up for large production of basic new types.

British fighter output behind the Vampire and Meteor designs still has considerable scope for development and may soon go to meet such various American basic needs, similar to that pushed by USAF, in the increasing need for jet fighters to increase range and endurance.

## Navy Inquiry

Little new turned up in probe of anti-B-36 campaign.

Navy court investigating sources of the B-36 "anonymous letter" seems to have more about Navy Capt. John G. Coombs' statement that the "navy is being added to death in the Foreigner."

The veteran Navy pilot and carrier commander will be called to testify before the court sometime Sept. 21 after a week's recess. Meanwhile, up-coming statements issued by both Navy

A Vampire and a Meteor were demonstrated equipped with afterburners on which the British call a reheat and another Meteor performed trials with two Avons. The \$5000 B. thrust behind the two fighters was evident in the performance of acceleration, climb and high speed passes on the deck.

► **Helicopters**—In addition to its highly rated Air Helicopter, Sikorski displayed a new version of its two-seater Sikorski helicopter that gave a modernized design touch to the basic Sikorski Mark I machine. The new Sikorski is priced at around \$14,000 and one is sold either as a personal helicopter or a trainer. Cockpit is fitted with dual controls, the cabin is pressurized from sea level up, allowing excellent view being in all directions, and an airtight glass panel transparent plastic is used for the main roof.

Powered by a Gipsy Major Ten engine, the Sikorski Mark II has a cruise speed on level speed of 85 mph and a 35-mph. cruising speed over a range of 180 miles. Absolute ceiling is 14,250 ft. Main rotor blades are plywood with all-aluminum tail rotor.

There were no new lightplanes demonstrated at the show.

It has been in Fleet Adm. William "Bill" Halsey (Ret.) and Rear Adm. Arthur K. Doyle, head of Navy Air Training, passed Coombs' "own dated outrage." Said Halsey: "He deserves the help and respect of all Naval officers."

► **Charge! Denial**—Coombs' charges that the Army General Staff dominated the armed services causing Navy fighting efficiency to suffer, and attacked the Tydings Law passed this year to tighten aviation in doing a broad toward disestablishment in the armed services.

His statement originally was prepared for an expected appearance at the Navy court, but when the court started without hearing him, he refused the

statement to the press. "The Navy's fighting spirit is going to get and I just can't stand it any longer," he told reporters. He expressed hope that his statement would bring on another congressional investigation, even at the risk of the court.

Ray L. C. Sasser (D., Md.), a top-ranking member of the House Armed Services Committee has called for congressional inquiry on the roles of Army, Navy and Air Force in defense and security, adding that differences between the services should be watched carefully so that "redirection of one branch of the service doesn't sap spirit, morale and destroy the spirit and morale of the other branches."

► **Naval Testimony**—Clair L. Martin, chairman of the company leasing his name testified earlier before the committee going as assistant of his consultant, such as George Wark, then assistant to John Nicholas Brown, Assistant Secretary of the Navy for Air. Worth admitted writing the "navy memorandum" in testimony before the House Armed Services Committee (Aviation Week, Aug. 25 and Sept. 2).

Martin testified that Harold Moser, his Washington representative, made an appointment for Worth to see him in his telephone office on Aug. 11. Worth, Clair Tom Davies, and Moser attended the conference.

"Mr. Worth said in substance, that he was greatly concerned with certain current developments in the aircraft field which had been and were then being reported or rumored in magazines and newspapers and elsewhere. He stated interest in seeing me was to find out whether I had any factual reference book bearing upon any of these reports or rumors," Martin said.

► **Tuned Controls**—"Only subject which was brought up on which I supplied any such information was the matter of the threatened cancellation of our best contracts during the war and the transfer of that business to the Eastern Electric Co. He had heard—had some—but there had been such an accident and I told me about it."

"I confirmed the fact that at one time such a termination and transfer had been seriously considered but had not been put through. That is the only bit of information, regarding as the document introduced Aug. 24, 1949, before the House Committee on Armed Services, which I could, as a state, be said to have supplied. That bit of information, such as it was, was correct."

► **Study Meeting**—Martin and Worth advised how he was compiling material on the reports and rumors, and later telephoned from Washington to offer a copy of his compilation. Martin was flying to Washington the following

## THE CITIES SERVICE FLIGHT TEST

*Power Prover*

This remarkable instrument indicates the combustion characteristics of any type of engine and atmospheric conditions in related engine parts.



### The Flight Test Power Prover



is typical of Cities Service's contribution to better flying. This instrument indicates combustion and atmospheric conditions in all types of engines and related equipment where atmospheric investigation are required. Through its use many engine's operating problems can be solved.

The Cities Service Flight Test Prover helps improve helicopter operation, resulting in smoother engine performance and better burning of fuel.

Cities Service has grown up with aviation. It has pioneered in many unique and unusual products and services for better flying. This ex-

perience stands solidly behind the complete line of top quality aviation products of Cities Service. Look for the green and white aviation emblem at more and more airports every day.

Cities Service Aviation Gasolines  
Cities Service Kerosene and  
Cities Service Aero Oils  
Cities Service Clean Solvent Engine Cleaner  
Cities Service Aero Oils and  
Aviation Specialty Lubricants



AVIATION PRODUCTS

New York • Chicago • In the South: Richmond, Fort Worth, Tex.



# THE SPARK OF LIFE FOR JETS...



Bendix-Scintilla's dual ignition systems are specified on military equipment as being the leading modern type of engine.



## ...AGAIN IT'S BENDIX-SCINTILLA!

The fact that Bendix-Scintilla Ignition Systems are used on some of the leading jet aircraft engines is noteworthy in two respects. It is powerful confirmation of their dependable, proven performance in all types of aircraft power plants, and it is a definite recognition of the vast experience,

manufacturing facilities, and resources which form the Scintilla Magneto organization. We believe the past and present performance records of Bendix-Scintilla Ignition Systems warrant primary consideration by aircraft engine manufacturers in every category.

**SCINTILLA MAGNETO DIVISION of  
SPERRY, INC.**  
Dept. 544, North Wabash Street, Chicago, Ill. 60604, New York 10012, New York



# AERONAUTICAL ENGINEERING

## Gust Loads Challenge Aircraft Designers

Present-day trends in airplane performance, weight, loading, size, bring major problem in gust resistance.

By Robert McLarnn

Most of the trends over the past ten years have aggravated the problems of aircraft design. These trends have all stressed their improved aircraft performance principally facilitated by improved powerplants. The tremendous power, both from turbojet and piston engines, have permitted substantial increases in speed, altitude, gross weight, wing loading and engine thrust, all tending to transfer greater energy to and absorb greater energy from the atmosphere.

The flight of today's airplane represents a much greater exchange of energy than did the flight of the average pre-war plane.

**Gust Load Tossing**—Each of these separate design trends has had an important effect on the relative susceptibility of the airplane to gust loads. Increase in cruising speed, for example, means that the airplane covers more miles per hour than did its predecessor and that its structure is exposed to more loads per hour.

An important trend has been a steady increase in useful load to gross weight ratio. Although this indicates improved structural design efficiency on the basis of strength/weight ratio, it also indicates a greater increase in working stress, both at a load factor of one (level) and at ultimate design load factor.

Therefore, because increased gross weight has resulted in a steady increase in design load factor which, in turn, implies an increase in working stress.

All of these trends, in the aggregate mean that the modern airplane is operating at greatly increased stresses and severity of loading simultaneously with lighter working stresses and lower load factors. Both of these conditions add up to grossing great loading troubles and there are becoming increasingly evident in accident reports.

**Artificial Load Factor**—An obvious solution is simply to remove the design load factor of the airplane to a point well beyond any loading encountered in flight. This approach actually has been taken in the case of the Bell X-1 and other research planes in the design of which an overall ultimate load factor

of 15 was arbitrarily selected in the hope that this would exceed any flight loads encountered.

To date this figure has proved adequate and there have been no instances of structural failure in research craft, even at supersonic speed.

However, this approach is both uneconomical and undesirable. It is uneconomical because the airplane's performance would be penalized severely through the added weight, which is necessary since this 50 percent of the true.

It is undesirable because, despite extensive work that has and is being accomplished, comparative data on maximum loads experienced in flight are as yet available.

The only sound approach is to determine the severity and frequency of occurrence of gust loads in structure, fatigue properties of aircraft materials, effects of stress-concentrating factors and effects of the general airplane design and performance characteristics.

The available approach can produce an airplane design of a limited, but known, life expectancy. And this is the most economical airplane to build and operate.

**Gust Cause**—Gusts are manifestations of turbulence created by factors both in the horizontal or vertical velocity of air currents. They come under all atmospheric conditions. Clear or cloudy days, when the wind is strong or light, at night or during the day.

Gusts caused by fluctuations in the horizontal movement of air temporarily alter the craft's speed with respect to the atmosphere, resulting in random upward or downward motion of the airplane.

Consider, for example, a plane flying with a tailwind. If this tailwind suddenly ceases, speed of the plane relative to the air is increased and the plane drops. Conversely, if the tailwind suddenly ceases, speed of the plane with respect to the atmosphere is increased and the plane suddenly rises.

In the case of an airplane flying into a headwind, a disturbance of wind velocity produces a slowing of the aircraft with respect to the air, and it drops as a consequence. An increase in the headwind produces a temporary increase in

plane speed and the craft rises upward in a "bump" all too familiar to those who fly.

**Range**—This same relationship obtains when the airplane climbs or descends through varying layers of air having different densities. There is a marked alteration of the atmosphere and when passing from a strata producing a steady velocity to one producing a lighter tailwind, the speed of the airplane is increased and a "bump" results. If moving into a strata with a stronger tailwind, the speed of the airplane is decreased and a "pucker" results.

This effect is reversed if the airplane is flying into a headwind.

Obviously, the gustiness produced by purely horizontal variation in wind velocity produces only a minor change in airplane lift and, therefore, only minor structural loads.

**Major Problem**—It is the vertical air movements that comprise the major gust problem since these reach velocities as high as 100 mph in thunderstorms of great violence.

These vertical air movements, both up and down, are caused by variations in the density of adjacent masses of air resulting from changing temperature conditions.

Generally, a south wind is warmer, the air has less density, and it rises and, when the two meet, the colder air forces the warmer air upward, resulting in a density gradient existing between the two.

If the density difference is pronounced, as in tropical areas, the vertical velocities within the front can be violent and exceedingly irregular.

Naturally, for any mass of air that rises, an equal mass must sink down, the sinking vertically ascending and descending air currents producing both positive and negative gusts.

However, downward moving masses usually travel some shorter area. Accordingly, these are less serious.

**Gust Effects**—As previously mentioned, gusts due to variations in horizontal velocity of the air cause changes in the speed of the airplane and therefore affect only the tail half of the wing but not the lift coefficient.

Vertical gusts directly affect the wing lift coefficient and, thus, coupled with their greater velocity, create loads not

ciently great to result in structural failure of the wing and consequent destruction of the plane.

Consider an airplane in steady level flight whose wing has the characteristics shown in Fig. 1(a). Now, assume that the plane, flying at velocity  $V$ , encounters a sharp upward gust with vertical velocity  $KU$ , as shown in Fig. 1(b). These horizontal and vertical velocity components produce an angle-of-attack increment of  $\Delta\alpha$ . If this angle-of-attack increment is added to the existing angle of attack, as shown in Fig. 1(b), the result is a lift coefficient increment of  $\Delta C_L$ .

Since the angle in radians is approximately equal to the tangent of the angle, for small angles,  $\Delta\alpha \approx KU/V$  and  $\Delta C_L = \text{slope} = K(U/V)/V$  in which  $V$  is the slope of the lift curve  $C_L$  per radian. The load factor placed on the airplane is an addition to its steady load factor and is, therefore, a load factor increment  $\Delta n$ .

From the foregoing, this may be expressed:

$$\Delta n = \frac{\Delta C_L}{C_L} = \frac{K(U/V)}{C_L} = \frac{K(U/V)}{C_L} = \frac{K(U/V)}{C_L}$$

in which  $K$  is gust alleviation factor (to be discussed later),  $U$ , gust velocity;  $dp/d\alpha$ , slope of lift curve;  $C_L$  per radian;  $\rho$ , density;  $S$ , wing area;  $q$ ,  $V$ , velocity of the airplane; equivalent airspeed;  $W$ , wing area;  $W$ , weight;  $W$ , weight;  $W$ , weight.

**Light Phases and Theory**—It is apparent that a time interval must elapse between the instant of application of the vertical gust velocity and the airplane change in angle of attack on the air plane wing, and for this reason the factor  $K$  is used to indicate the relative alleviation of the gust effect by various air and/or airplane characteristics.

For example, a lightly loaded air plane of the personal type would be quickly affected by the gust and ride upward fast enough to dissipate most of the gust force and result in little or no increment in wing loading could result.

A heavily loaded plane, on the other hand, would absorb most or all of the gust as an increment in wing loading with little or no alleviation of its effect.

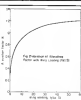
Fig. 2 illustrates this variation with alleviation factor with airplane wing loading, and shows that the factor is quite low for low wing loadings but increases rapidly for wing loadings above 16 lb./sq. ft.

Effect of wing loadings higher than about 50 lb./sq. ft. on the alleviation factor is largely unimportant.

**NACA Studies**—One of the first design questions involved in a study of gust load criteria is the magnitude of the gusts the airplane may be expected



Fig. 1—Effect on Airplane Lift Coefficient of Airplane Flying Past (Ref. 4)



to encounter. The airplane cannot possibly be designed to withstand the maximum gust velocity of record, nor even the maximum estimated to be expected in service.

In establishing a gust load criterion, it is necessary to determine the highest gust velocity normally encountered in the service for which the airplane is intended, as the rate of transport aircraft, angular airspeed.

To determine this criterion the National Advisory Committee for Aeronautics undertook a statistical study of airline operations in 1952 by the installation of V-G recorder equipment on transport craft then in operation. These included American Airways transport Fokker F-50s operating between Cleveland and Dallas, and Pan-American Convair 440s on the Ford 5A7-Cs operating between Boston and San Francisco, and Douglas DC-3s on a variety of routes.

Subsequently, recorder units were installed on the Boeing 307 Strato-liner, Boeing 314 Flying Boat, Martin 130 Flying Boat, Sikorsky S-47 flying boat, Lockheed NC 35 high-altitude transport plane, and the Boeing XB 15 bomber

World-Wide Recording-V-G Recorder installations have continued in all the modern transports, both land and sea, so that a continuous and comprehensive record is being obtained for a wide range of aircraft designs and operating conditions.

In addition to these routine operational records, several Northrop F-6B Black Widow two-engine fighters have served as specially-equipped research planes, flying directly through violent thunderstorms in the U.S. Weather Bureau Thunderstorm Project.

Measuring both records are being obtained by both Air Force and Naval Airbases, although these data only indirectly indicate effect of gust loads.

East Magnitude—Most severely published data cover only those transport records taken from 1932 to 1942 and do not include information on the more heavily loaded and faster postwar transports.

However, these records are indicative of the general magnitude of gust velocities which, of course, do not vary with airplane size or geometry. Since they cover 74,820 lb. of loadplane operations, and 66,129 lb. of flying boats, it can be expected that they furnish an adequate cross-section of gust characteristics along the entire flow.

These data indicate that the maximum effective gust velocity, based upon airplane design wing loading, normally to be encountered in service is 30-40 ft./sec. at all useful operational speed values.

Although at first glance it is indicated that this value is well in excess of the standard design value of 30 ft./sec. points are to be noted.

Value of 30 ft./sec. must be multiplied by  $K$  which, in the case of most modern transports, brings it up to approximately 50.

The value of 30-40 ft./sec. is based upon the airplane design wing loading, which is in excess of the average operating wing loading at the bulk of the records.

It is noted, moreover, that the wing loading in the bulk of the records was only 35 percent of design, which would further reduce the maximum effective gust value most ready to that used in design.

Frequency of Occurrence—Having determined maximum normal gust intensity, frequency of occurrence of gust intensities cannot be determined if a proper evaluation of aircraft life expectancy is to be made. Frequency of occurrence of gusts is determined largely on the value of the threshold wind, that is, the smallest gust velocity which the instrument is capable of recording.

If the instrument only recorded gusts greater than 20 ft./sec. and those at three crosses in 1,000 ft. of recording, then it would be assumed that one gust occurred each 400 ft. while if the instru-



The Douglas AD-2 is another of the famous aircraft — both conventional and jet-propelled — which rely on Clifford Feather-Weight All-Aluminum Oil Coolers.

Superior strength-weight ratio derived from Clifford's patented method of forming aluminum and accurate performance ratings derived in the Clifford wind tunnel laboratory — the best and most modern in the structural heat exchanger industry — account for the rapidly growing acceptance of Feather-Weight Oil Coolers.

Your engineer is also invited, CLIFFORD MANUFACTURING COMPANY, 136 Grove St., Waltham 24, Mass. Division of Cleveland-Tennesson Corporation, 410 W. New York, Detroit, Chicago and Los Angeles.



ment showed gusts of 1 gpc and 1200 lb. of force, thus a value of one gpc per hour would result.

The NACA V-G recorder used in most of these tests had a threshold of 0.5 gpc and cannot resolve minute gusts of much lower velocity.

Extensive statistical analysis is required for a meaningful interpretation of the mass data. For example, only those records taken from aircraft operating under approximately the same conditions of kinetic weather, speed and altitude are comparable. "Clean intervals" must be selected carefully so that all of the data may be used in plots of frequency vs. velocity. There is a variety of methods used in the analysis of such records and therefore he is selected with great care and objective consideration be reached.

Fig. 1 gives the results of the study in a plot of relative frequency vs. effective gust velocity. The graph includes a relative distribution which characterizes several widely divergent results obtained under various conditions. For example, one of the worst gusts ever encountered in piston twin-engine operations. There is also shown the relative independence of gust frequency from airplane size, since they include data on aircraft ranging from an Aerojet C-1, used in a special investigation, to the 55 ft long Boeing B-74 flying boat.

• **Peak Ratio**—The fact that gust frequency distribution follows a fixed peak ratio indicates that the total frequency is proportional to the distance flown within turbulent regions. To determine the frequency of the independent gusts, an average gust interval,  $\bar{t}$ , is used and defined in the average distance along the flight path in turbulent air between significant gusts. Plots of this value vs. altitude, wing chord showed an approximate linear relationship between the two at a value of 13 chord lengths, and this value is used in the average gust interval.

It is shown that transport flying conditions during which significant gusts are encountered, followed by entry into turbulence where significant gusts are encountered with great frequency. Relationship between these two types of flight condition is termed the peak ratio and is defined in the equation:  $P = 5193 R_L / (V \bar{t})$ , in which  $P$  is total frequency,  $R_L$  path ratio,  $\bar{t}$ , length of flight between, sec, and  $\bar{c}$ , chord length, ft.

With ratio and, therefore, total gust frequency for any particular flight, will depend on the operating conditions. For example, a freeline transport operating overhead at low altitude would encounter a greater percentage of turbulence as did an airplane operating at high altitude above the mechanical

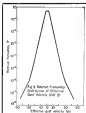


Fig. 1 Relative frequency distribution of effective gust velocity

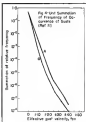


Fig. 2 Peak ratio relationship of frequency of turbulence at different altitudes

turbulence of the ground and above that of the convective clouds. On the basis of available data, an average value of path ratio of 0.1 appears reasonable.

From the foregoing, then, it has been determined that the transport airplane will encounter gusts at a least 10 percent of its route or time and that within that gusts at it will strike a significant gust every 13 chord lengths.

• **Stress Proportionality**—It is now pertinent to determine the frequency of occurrence of related loads on the air frame structure. In choosing a gust frequency for design purposes, the choice may be between a distribution having relatively high frequency at the higher values of effective gust velocity, or one having higher frequency at the low effective gust velocities.

Fig. 4 indicates that two limiting relative-frequency polygons approximating the approximate upper and lower limits of the data used in Fig. 1. This is the basic design chart used in deter-

mining values of stress frequency. However, there are several phenomena that must be included in the application of these data to stress analysis.

Analysis of V-G records on the basis of simply counting the "peaks" minimizes the possibilities of superposition of small gusts on the larger gusts counted. In determining the number of gusts encountered in a flight, only those peaks which began and ended at the IG datum line were included, since they most accurately indicate a gust.

However, a special count of all peaks, regardless of point of origin, indicated that the number of such superimposed peaks was about twice as high as those occurring independently about the datum line. Most of these peaks occurred near the threshold value of 0.5 gpc.

Although these are not of practical significance in a determination of major loads on the structure, they may be of significant importance in determining airframe life.

• **Specific Gust Data—Distribution of the gust velocity is, of course, not always uniform along the span, and on asymmetrical gusts will occur with a frequency depending on wing planform and wing arrangement.** Random, asymmetrical gust load stress can not be avoided, although some work has been done based upon specific airplanes.

Using data obtained from the Boeing XC-15, a semi-symmetrical criterion has been developed. While not applicable to the design of the main aircraft structure, it is of value in the design of engine mounts or such large mass distributed in a fuselage from the wing root out, therefore, subject to asymmetrical gust loads.

This criterion establishes an effective gust velocity at the tips of 20 gpc, positive at one tip and negative at the other, with a linear distribution between the two. It states  $U_{eff} = 0.5 U_{tip}$ , in which  $U_{eff}$  is symmetrical effective gust component to be used concurrently with the conventional component, and  $U_{tip}$  is the design symmetrical effective gust.

Data based on results of the XC-15 Gust Research Project, however, indicate that this criterion is conservative, since the XC-15 data show the maximum amount of effective gust velocity obtained in flight for the unsymmetrical gust was 1.5 gpc. These tests also indicated that the most probable lateral gust shape encountered in flight is triangular, with the other gust shapes, in order of their respective frequencies, being (1) the unsymmetrical and (2) the rectangular combined with the trapezoidal shape in which the effective gust velocity is constant across the span to one tip but drops sharply at the opposite tip.

If the magnitude of the average of lateral gust velocity for the critical in-flight gust is considered to be 10K ft/sec, the average effective gust velocity for the critical unsymmetrical case, based on the same frequency, would be 21K ft/sec. Lateral gust gradient distance was found to be a value of 9 chord lengths.

However, all of these data are of preliminary nature only and considerable work, both theoretical and experimental, remains to be done before a rational criterion, such as for design, is available.

• **Dynamic Response**—A third phenomenon affecting the direct applicability of Fig. 4 is the dynamic response of the structure. Gust loads for simplicity of calculation, have previously been assumed to be static loads applied to a rigid airplane.

For personal type aircraft, the assumption of a rigid structure appears to be reasonable but the trend towards larger and heavier airplanes necessitates the determination of methods for the reduction and evaluation of the effects of dynamic response.

It is in this aspect that gust loading takes on the aspects of an aerodynamic problem.

Several approaches to this problem have been taken, each with underlying assumptions. Proper analysis of the loads, mental models of bending of the wing under symmetrical loads which are assumed to be known by replacing the airplane with an equivalent airplane and elastic system.

The problem is then resolved into one of obtaining the proper constants to be used as coefficients in equations on linear differential equations which represent the equivalent system. This method is supported by experimental tests.

It has indicated that the dynamic stress ratio for airplane wings increases as the gradient distance of the gust decreases, in comparison of a 13-chord gradient distance leads to as much as a 12-percent variation and repeated gusts do not seem to be more critical than a single gust.

In a study of changes of airplane gust vectors, the method indicates that a change in forward velocity does not appreciably change the dynamic stress ratio, a reduction in wing frequency by a change either in weight or in stiffness of the wing results in an increase in the dynamic stress ratio, wing tip acceleration increases a generally much greater than the factor increases from movement and the ratio of the two movements tends to increase in the speed increase. Other calculations show that this ratio is about two at a speed of only about 200 mph.

(Continued on page 12)

## engineer's notebook



### FOR HANDLING SHOCK LOADS

Secure overflow cylinder to loading gear slings dampener with Marmen T-bolt dampers, especially suited to severe shock loads encountered in this installation.

### JET ENGINE JOINT CONNECTIONS

Marmen V-Bond couplings provide an efficient seal and facilitate maintenance with quick disconnect provisions of the coupling latch.



### FASTENING OXYGEN BOTTLES

Holding straps and dampers prevent whiplash a rebound force in any direction of 3500 lbs. Standard Marmen types fulfill all requirements and can be adapted to any specific design.

Save cost and design time with Marmen

FOR INFORMATION WRITE DEPT. W-9  
OF THE ADVERTISING AND COMPASS

**MARMAN**  
PRODUCTS CO., INC.

140 WEST FIDELITY AVE.  
INGLEWOOD, CALIF.

**something  
big  
is  
happening  
in the sky**

If you're looking for new markets to keep your sales volume up—  
*look to the sky!*

If you're interested in getting your share of a 5 billion dollar market—  
*look to the sky!*

If you'd like to latch on to the most exciting, fastest-growing industry  
in America today—*look to the sky!*

The aviation industry is not just big—it's tremendous! And it's  
growing every day.

Right now, it's spending at the rate of almost 10 million dollars  
a day. More than half of its material requirements will come from  
subcontractors and secondary suppliers—and that means people  
like you!

Look over the following items carefully (there are many others),  
and see if they don't suggest sales opportunities to you:

Electronics and photographic equipments, special purpose motors,  
wires on end of wiring and cable, magnesium, steel, copper, rubber,  
seals, bearings, hardware, tools, crating systems and devices,  
trucks, cars and fasteners...in fact, countless other products of  
almost every known industry.

For complete details, write for your copy of "Aviation Week—and  
the Market it Serves." Aviation Week, Dept. B, New York.

McGraw-Hill Publishing Co., Inc., 330 W. 42nd St., New York 36,  
N. Y. . . . Offices in Boston, Philadelphia, Pittsburgh, Cleveland,  
Detroit, Chicago, St. Louis, Dallas, Atlanta, San Francisco and Los  
Angeles.

*Look to the Sky*

*for your market*



**Another Study**—A similar study by Sells<sup>1</sup> which covers a wing clastic on leading but rigid to torsion, said an anelasticity with springs supplying the elastic properties and two-dimensional flow conditions causing over a typical wing.

These results indicate that torsionally the aerodynamic forms damp out the oscillation at high speeds, so that stresses greater than steady-state values do not occur at these speeds.

However, the results are applicable only to the case of a wing subjected rigid to torsion, and these calculations would have to be extended to include the relative oscillations in bending and torsion of an actual wing.

Because the natural period of wing sections almost in direct proportion to the wing dimensions, and because the size of gusts to which the airplane will respond increases as the airplane size, the rate of natural period to period of application of load means constant for equivalent flight speed. Dynamic response of the structure would, therefore, appear not to increase with airplane size.

Probability of structural failure from gust loads increases rapidly with overloading of the airplane.<sup>2</sup> Total applied load from which an airplane structure is designed is composed of a steady load equal to the weight of the airplane plus a load allowance due to maneuvers in flight.

An increase in the steady load is direct for allowable moment due to gusts. Effect of air overload on critical gust frequency is shown in Fig. 5 and indicates the occurrence of such situations.

As a result of Civil Aeronautics Board concern with the importance of gust load criteria as a major transport design factor, design peak load severity has increased steadily over the past few years.

Civil Air Regulations, Part 25, specify



Fig. 5—Variation of Critical Gust Load Frequency with Airspeed for Typical Wing

for 30 ft/sec. Part 4b specifies 40K ft/sec, and proposed Part 25 further increases this value to 50 ft/sec.

**Effect on Fatigue Life**—Effect of gust loads on an airplane structure is of two aspects: compression. Designers previously have been concerned principally with the first of these effects—moment of loads which exceed the structural strength of the airplane, resulting in failure.

The relation in this problem has been to determine how strongly and how often such gusts occur and to design according to these data.

The second, perhaps more important, but much more obscure effect of gust loads, is a reduction in the life expectancy of the airplane through fatigue. In this case it is the cumulative effect of the small parts, rather than the load applied by a few large gusts, that particular designers desired to be taken into account.

For example, it is common practice for pilots to reduce speed when entering turbulent air, to reduce the load factor increment caused by the gust. Oddly enough, however, this slowing down has a negligible effect on the fatigue life because only the stresses resulting from the relatively few large gusts and from a

small number of the smaller gusts have been diminished.

Fig. 6 presents a typical "S-N" curve, which illustrates the well-known condition that the number of cycles required to cause failure decreases as the stress amplitude or moment stress increases.

The fact that slowing down of the airplane to study its low lift effect on fatigue life follows from the relation, shown in Fig. 6, that diminishing the subsequent large stresses on the wing surface does not influence the number of cycles of the smaller stresses required to cause failure of the structure and S-N curve.

Another paradoxical relationship between gust loads and fatigue life is that, although higher wing loadings suggest the structural failure risk of gusts, the net effect of higher wing loadings on the fatigue life is favorable, since the favorable effect of reduction in stress amplitude has more than offset the deleterious effect of the increase in mean stress.

Effect of increasing speeds has only a moderate effect on fatigue life, since increasing speed approximately which the favorable effect (caused by increasing the wing loading). Hence, if speed and the wing loading are increased concurrently (which usually follows), not much change in fatigue life occurs.

#### References

1. W. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
2. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
3. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
4. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
5. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
6. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
7. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
8. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
9. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
10. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
11. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.
12. S. Sells, "Fatigue Life and Structural Load," *Proc. 10th Annual Meeting of the American Society of Mechanical Engineers*, 1943, New York, N.Y., 1943.

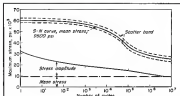


Fig. 6—Sample Data for Computation of Fatigue Life (Part 1)

## Northrop Acquires Electronic "Brain"

First private firm to purchase an all-electronic computer to help solve its engineering problems is Northrop Aircraft, Inc., Hawthorne, Calif.

Called BINAC, the machine was built to order by Northrop for Edwards-Mausly Corp., Philadelphia, and reportedly is the second of the only two privately financed "brain" units in existence.

Only similar computer is its much larger 50-ton unit, ENIAC, developed in 1946 for the Army by inventors, J. W. Mauchly and I. Z. Zuse. With dimensions of only 5 x 4 x 1 ft., the new computer fits in an average size office and is tiny in comparison to its predecessor.

BINAC actually consists of two components having electronic switches for quick checking at every step in the solution of a problem. Also included are two memory tape units, which are the main reason for BINAC's lightweight. Invented by Edvard, these units make it possible for the computer to get along on 1400 vacuum tubes—most of the 15,000 required in the ENIAC.

For the past month, Northrop engineers and mathematicians who will operate the machine, have been put through an intensive training course by Edwards-Mausly. It is estimated the computer will assist at several operations and server centers, solve two mathematical problems, and typists to perform arithmetic operations.

According to John K. Naughton, president of the aircraft firm, "BINAC will be of considerable value in reducing preliminary design and test time on most research and development projects... calculate fairly impossible or impossible of solution can be completed readily with savings of hundreds of thousands or possibly even millions of dollars in time and money."

## Steel Alloy Developed For Low Temperatures

A new alloy that, reported capable of making casting engineering in equipment at temperatures down to -823 F., has been developed by Libby Steel Foundry, Libby, Pa.

An austenitic cast iron alloy, Grade 25, containing, in part, 11 and 9 percent chromium and nickel respectively, the material already has been used in production of steel castings for pressure equipment in storage facilities for liquid oxygen, used in

with **HANSEN** couplings

**minutes Saved**

are

**HOURS GAINED**

**CONNECTION**

**TAKES ONLY A SECOND**

If you have fluid lines in your plant you can increase efficiency and save precious minutes with Hansen Couplings. Any size can be quickly and easily connected or disconnected without loss of liquid or gas in the line.

When you want to correct a Hansen Coupling, it is usually necessary to push the plug into place, where it locks accurately. The valve opens, and flow resumes instantly. The same operation disconnects—move the plug out, and the plug pops out. Flow is shut off automatically.

With a wide range of available sizes and types you can select one suitable for any application. Air, oil, and gas, oxygen and acetylene, gasoline, hydraulic oil, and steam... make every type of service has a Hansen Coupling constructed to meet its specific requirements.

Write for Industrial Catalog No. 47 today.

**QUICK-CONNECTIVE COUPLINGS**

**ONE AIR • ACRYLUM • OIL • GAS**

**STEAM • PETROLEUM • HYDRAULIC**

**THE HANSEN MANUFACTURING COMPANY**

4031 WEST 150th STREET • CLEVELAND 11, OHIO

# IT TAKES ALL THIS to Type Test a Pesco High-Pressure Fuel Pump

...pump enough gasoline to fill 150 tank  
cars at enough pressure to fill a stand-  
pipe 2,000 feet high

1,500,000 gallons of gasoline ... pumped at 800 p.s.i. ... at  
temperatures ranging from  $-67^{\circ}$  to  $+385^{\circ}\text{F}$  ... at stress  
plane pressures that simulate the altitudes which jet planes  
must fly ... for 500 continuous hours ... that's the type test  
passed ... successfully ... by Pesco High-Pressure Fuel  
Pumps

Pretty rugged test! Particularly when you consider that  
pneumatics has no lubricating properties ... and no external  
lubrication is permitted.

Pesco High-Pressure Fuel Pumps are designed and pro-  
cessed built to meet all the exacting requirements of jet en-  
gine operation. Though light in weight and small in size they  
feature "Pressure Loading". Pesco's exclusive, patented  
principle of gear pump construction that makes possible high  
volumetric efficiencies over a long service life regardless of  
wide extremes in temperature and altitudes.

High-Pressure Fuel Pumps are one of many vital jet and  
reciprocating engine and aircraft accessories that have been  
developed by Pesco. Write for complete information.



**PRODUCTS DIVISION**  
BORG-WARNER CORPORATION  
14001 South Blvd., Chicago 5, Illinois

rocket engine. The molten is con-  
tained under high pressures at approx-  
imately  $-258^{\circ}\text{F}$ .

Recent tests on the new steel re-  
portedly have revealed no structural  
changes down to  $-423^{\circ}\text{F}$ , but have  
indicated appreciable increase in ten-  
sion directly proportional to dropping  
temperatures.

Break strength of the composite al-  
loy steel increases with temperatures,  
and in direct relation with hardness.  
Ductility is lowered, but not in propor-  
tion to increase in tensile strength,  
estimated to be about 265,000 psi at  
 $-423^{\circ}\text{F}$ . The steel is used to retain  
adequate strength factor and resistance  
to embrittlement at this temperature.

## Doman Demonstrates Rotor Stability

The Doman helicopter has demon-  
strated "unusual stability," in test  
flights lasting almost an hour, without  
the pilot touching one of the controls,  
according to reports from the center.

Despite windy conditions in a recent  
demonstration flight for the navy, the  
Al Bell, test pilot for Doman Helicopters,  
Inc., Danbury, Conn., reported the  
control stick, cyclic control and throttle  
for extensive periods of low vibration by  
riding during the trial.

After flying hands off for 45 min. in  
sunburn light, the pilot stated he could  
have continued in this manner until  
the fuel ran out.

The company attributes the craft's  
exceptional stability to "several basic  
jet pump improvements in the rotor  
and in the control system," which  
dissipate the wind for gyro devices.

A breakdown in parts also  
has been gained through elimination  
of complicated hinges and dampers from  
the four linked, oppositely balanced  
rotors.

First Doman production helicopter is  
called to be the "Doman," a 245-lb.,  
utility machine designed to carry 7  
passengers, at 1000 ft. of altitude for  
prolonged periods in flight.

## Dural Bars Fitted To Steel Jigs

Maintaining close tolerances in fab-  
rication of aluminum alloy parts always  
has been a tough problem in the air-  
craft industry.

This is especially true where alu-  
minum is welded in steel jigs, as areas  
subjected to fairly wide temperature  
changes. Dural, for instance shrinks  
or expands at twice the rate of steel in  
response to identical temperature  
changes.

If nothing were done to counteract

this expansion and contraction, com-  
ponents of the airplane fabricated in  
steel jigs at different times and tempera-  
tures would either gap or overlap when  
joined together.

While this problem can be overcome  
by switching from steel to dural jigs,  
costs and lack of material often pre-  
sents this trip.

A way out of this dilemma was found  
by engineers at Boeing Aircraft Co.,  
Seattle, Wash., through use of dural  
length bars fitted within the steel jig  
framework.

The length bars are attached to the

base of the jig through slots which  
permit the bars to slide just enough  
to compensate for the different ex-  
pansion-contraction rate between steel  
and dural.

The aluminum alloy pad or other  
component is worked on by the jig in the  
usual manner. But the dural bars are  
used to locate the jig members which  
control the length of the part under  
fabrication.

Being in the form however, that this  
compensation of steel and dural does not  
set all straight. It was all right for  
the same project, regardless of steel

**BARBER COLMAN**

**SENSITIVE POLARIZED RELAY**  
FOR D. C. CONTROL CIRCUITS

**Micropositioner**

OPERATES ON VOLTAGE  
OR POTENTIAL CURRENT

1. Precision control of relay operation  
2. No moving parts  
3. No electrical contacts  
4. No mechanical wear  
5. No electrical noise  
6. No electrical interference  
7. No electrical leakage  
8. No electrical arcing  
9. No electrical sparking  
10. No electrical burning

1. Precision control of relay operation  
2. No moving parts  
3. No electrical contacts  
4. No mechanical wear  
5. No electrical noise  
6. No electrical interference  
7. No electrical leakage  
8. No electrical arcing  
9. No electrical sparking  
10. No electrical burning

An extremely sensitive polarized relay used in numerous  
automatic control applications. Made of stainless steel,  
resistant to corrosion. Operates on 100 cps. Various models include  
relaying and terminal block bases, and metal plug-in  
bases open or hermetically sealed. Precision of operation  
is a leading characteristic which makes one side of the other  
of 5000 potential relaying with minimum mechanical damage  
and type is activated by polarity of applied voltage, flexible  
and type by differential of two currents. Radio appli-  
cations include remote positioning, synchronization, tem-  
perature control, and detection and amplification cir-  
cuits. Barber-Colman Micropositioners meet Army-Navy  
specifications. Write today for new Bulletin F-101  
giving complete data and suggested uses. Consult your  
Barber-Colman representative for any engineering ques-  
tions needed.

**BARBER-COLMAN COMPANY • ROCKFORD, ILLINOIS**





OPEN SOON . . . NEW AUTO-LITE WIRE AND CABLE PLANT  
at Hazelton, Pennsylvania. Over 750,000 square feet of floor space equipped  
with the most modern facilities for the production of wire and cable.



# AUTO-LITE

announces...

## INCREASED PRODUCTION... Aircraft Wire & Cable

Aircraft engineers! Designers! Executives! The tremendous output of this new Hazelton plant combined with the new expanded facilities at the Auto-Lite Port Huron plant makes it possible to meet additional aircraft wire and cable demands. The quality of these outstanding products is the result of 38 years of experience, research and

advanced laboratory tests. The specifying of Auto-Lite wire and cable is fast becoming a recognized practice among leading aircraft manufacturers who have found that money cannot buy better wire and cable. For an informative catalog, write to THE ELECTRIC AUTO-LITE COMPANY

Wire and Cable Division

Port Huron

Hazelton

### LOW TENSION

Aircraft wire with copper conductor  
Specification: AWG 30, 36

Aircraft wire with aluminum conductor

Specification: AMS 98

Shielded aircraft wire

Specification: AMS 128

### HIGH TENSION

Aircraft ignition coils with shielded coil  
conductors and magnetic shield  
Specification: SAE 1216, 1217, 1218, 1219, 1220 and 1221

Aircraft ignition coils with (1) shielded coil  
(2) shielded coil conductor and shield and  
magnetic shield  
Specification: AMS 1216

Aircraft ignition coils with copper conductor  
and shield and magnetic shield  
Specification: AMS 1216 and AMS 1217

Aircraft ignition coils with copper conductor and  
magnetic shield in commercial specification

Aircraft ignition coils with copper conductor  
and shield and magnetic shield in commercial  
specification

TUNE IN "SUSPENSE" . . . CBS RADIO NETWORK THURSDAYS . . . CBS TELEVISION TUESDAYS

## AIR TRANSPORT



LEASING PLANTS to stretched-out airlines, and performing maintenance on them, has kept California Eastern afloat in business.

### Coach vs. Cargo As Profit Maker

California Eastern, loser of \$900,000 as a freight carrier, pulling out of red in the air coach field.

Money-making potentialities of airfreight don't approach those of an coach in the opinion of one uncommitted expert who has tried both.

The carrier, California Eastern Airways, leased immediate liquidation last week than a year ago after losing \$900,000 in the freight business. Now it is slowly working away its debts through operations in the lucrative passenger field.

The striking reversal in CEA's fortunes is explained by company president Andre de Saint-Pierre this way:

► **Heavy Losses.** "Airfreight moves at about 15 cents a ton mile and coaches principally with freight and express which costs about 30 cents a ton mile. The carrier, therefore, pays a premium of about 50 percent for the greater speed of airfreight, which provides no real economy, delivery from coast to coast, compared with freight and fish, meaning delivery by the fastest surface transportation."

"That at present airfreight rates, an extremely high load factor is required to attain a break-even point. None of the major all-coast lines has been able to maintain such a high load factor, and, as a result, they have consistently reported operating losses."

By contrast, since a passenger and his baggage weigh, on the average, 200 lb., the air coach rate of 4 cents a ton-mile is equal to 40 cents a ton-mile. The potential revenue per plane mile in air coach is, therefore, 250 percent higher than in air freight.

"It is probable that with the progress of aircraft design, freight will move by air in volume at rates more competitive with surface transportation at some time in the future. Such improved air rates, however, will also compete even more successfully against surface rates in passenger transportation."

► **Raised Problem.** "Reason for this is that the cost to the airlines of loading passengers representing a weight of one ton is at least three times greater than the cost of loading one ton of freight. The cost to stage an coach line at 4 cents a passenger mile is 50%, compared with \$75-82 for rail coach and \$40-50 for bus."

"An coach takes 57 hours less time coast to coast than rail coach and 51 hours less than bus. Further, the passenger obtains a substantial saving in mail costs."

Saint-Pierre recently told the Senate Interstate and Foreign Commerce committee that profitable and unobscured transcontinental air coach service at \$78 a mile is feasible with modern equipment.

► **Reversal Reported.** By leasing its four DC-4s to large freighter carriers, California Eastern raked up a \$100,000 net profit during the last half of 1949. The Oakland, Calif., company paid off its priority creditors in full, with interest, and has high hopes of becoming in the black.

During the last part of 1949, California Eastern was the third largest independent all-coast carrier, ranking be-



SAINT-PIERRE: Heeds you win, cargo you lose.

head Sherk Airways and the Flying Tiger Line in two miles flown. But in May of last year CEA went into bankruptcy with its application for an airfreight certificate still pending before the Civil Aeronautics Board.

► **Cargo to Coach.** The company had the choice of quitting business with the equity of its stockholders being wiped out or finding a new source of revenues that would extend operations. In June, 1949, CEA decided to convert one of its C-54s to passenger service as an experimental lease. (It now is converting all four of its C-54s to 71-75 seats.)

The plane was leased to a non-scheduled air coach operator in July, 1949. Since then, the company has made steady progress back to solvency. Originally it leased planes, tires and facilities to non-scheduled lines, and pas-

# Snap-on

## Open End Wrenches

with the famous "Blue-Point Supreme" Trade Mark

...are unequalled for strength, toughness and fatigue-resistance



6-PC. Wrench Set

Twelve popular wrench sizes are 4 handle, from 3/16" to 1 1/2". No duplication of sizes.

In leatherman bag . . . . \$8.50

Yield only . . . . . \$7.70



Sturdy 9-PC. Wrench Set

12 wrench sizes made in more popular sizes displayed on different handles. Sizes from 3/16" to 1 1/2".

In leatherman bag . . . . \$12.75

Yield only . . . . . \$11.10



Popular 9-PC. Wrench Set

Eleven in size of Snap-on's most popular Open End Wrench Size. Effects are 12 different wrench sizes also shown from 3/16" to 1 1/2".

In leatherman bag . . . . \$16.15

Yield only . . . . . \$14.50



Complete 12-PC. Wrench Set

Full range of sizes from 3/16" to 1 1/2" with 7 open end wrench sizes displayed on different handles.

In leatherman bag . . . . \$17.00

Yield only . . . . . \$15.35

**Snap-on Tools**  
THE QUALITY OF 100 YEARS

1918 THIS COULD TODAY  
**SNAP-ON TOOLS CORPORATION**  
2024 S. 24th Ave., Kenosha, Wisconsin

Delivered Ex-Factory (Check) (Money Order or) (Company Purchase Order for Open End Wrench Set) delivered to you:  
\$50.00 \$64.12 \$20.72 \$2.58  
\$102.32 \$116.44 \$41.14 \$4.98

With additional service charge for delivery at Snap-on's factory for your State or City.

Name \_\_\_\_\_  
Address \_\_\_\_\_  
City \_\_\_\_\_ State \_\_\_\_\_

Snap-on Tools are the standard only from Snap-on. Beware of other offers and hold only one Snap-on Set.

forced all the maintenance. Then he took CAB and that arrangement was a solution of the live, CEA followed its pilots and the live engine for equipment based shops. That made everything legal, CEA says.

A month ago, California Eastern's application for an all-cargo certificate was turned down by CAA. But CEA does not own Shio, the Flying Tiger and U.S. Airlines, which retained sole rights. In Street-Pfleger's opinion, the status of the flight certificate remains uncertain "and that operators are believed to contract accordingly."

►Expense Considered—But CEA also is realistic about its own future. It wants to broaden its operations so that besides leasing and maintaining equipment it may engage directly in wide-scale flying operations, in regular interstate and regular intrastate air coach transportation, and possibly in scheduled domestic coach service if CAA grants its application for a passenger-carrying certificate.

California Eastern believes that in time it would own its four DC-4s and other and earn a satisfactory profit in its revised capital if it could insure adequate continuance of the company's present assets from leasing equipment to nonaffiliated operators. Fundamentally, it thinks that leasing aircraft to carriers which do not possess sufficient capital to own all the equipment necessary for these requirements can be a sound business.

By leasing equipment to different users having complementary peak demands, CEA could supply the requirements of a number of lines and keep its wrench here. Actually, however, the problem isn't that simple. Some of California Eastern's customers may eventually be forced out of business by their inability to comply with CAA's accelerated regulations. Others may be able to get out and purchase their own planes.

►Certificate: Etc.—Consequently, CEA last April filed its application for a noncommercial coach-type certificate. Reule reports of other companies are to be heard by CAA at the same time, and no decision in the case is expected before late next year. Meanwhile, California Eastern wants to conduct air coach operations on its own hook.

CEA is especially proud of its maintenance shop at Oakland. It claims the ratio of employees to planes is better than that of any DC-4 in the service of any low-engine operator in the U.S.

The company hopes to secure additional outside maintenance work from the Air Force and from other carriers. At present, CEA has only a small amount of outside maintenance business.



STRATOCRUISER COCKPIT looking aft; it was version of Dr. Debel's simulator



AT-4 MOCKUP is another, showing the side view of application for the system.

## How to Train Crews at a Profit

Delham electronic trainer, as used by Pan American, frees aircraft for revenue producing operations.

By Stanley L. Colburn

Dr. Richard C. Deham's wonder box—the electronic flight simulator—now nearly as much as a twin-engine medium transport, and it doesn't carry

any payload. But operators are finding the simulator is industry's top-dollar—can produce a profit in training pilots.

At Pan American Airways, where the simulator is a complete replica of

the Stratocruiser cockpit, pilots are putting in 75 percent less time in the actual aircraft. Training flights in the 800-horsepower would cost nearly \$1500 per hour with the possibility that a serious accident on the part of the crew could cost the line as much as \$1.5 million.

►Lower the simulator costs. As an operating cost of \$50 per hour it leaves little in the aircraft and crew, at the same time eliminating the need of taking a revenue-producing transport out of service.

Curtis Wright, manufacturers of the device, estimates it can be used to replace 90 percent of training flights necessary to maintain flight crew proficiency in simulated emergency situations in actual flight. C.W. also says the simulator can solve by a two-third aerial navigation flight training in specific routes.

►Economics—Here's what it adds up to in dollars and cents: Pan American's simulated five-stratocruiser pilot (using on the actual Stratocruiser) is 10.5 in per pilot.

Based on \$1000 per hour, this would be \$10,500. Not including the Delham simulator, PAA estimates simulator training will be 15 in per pilot in the taking and 4.95 in the air. Taking the figure of \$50 per hour for the simulator, and \$1800 per hour for actual flight, this would be \$4830, or a saving of \$5770 per pilot in Stratocruiser training.

While training cost figures vary for different airlines and aircraft, it is reported that the Delham flight simulator—which costs between \$200,000 and \$300,000—saves more training at less cost to the operator. Carriers now wonder what other flight problems can be solved by mechanical simulation of conditions.

Pan American Airways, so far the only airline purchaser of a simulator, is helping pay the purchase cost by training Stratocruiser crews for British Overseas Airways Corp. and American Overseas Airways. But airlines are not the only users of the Delham simulator.

An F-100 is in the process of using simulator to train service pilots in six types of aircraft, including the B-50. Curtis Wright has orders for more than 40 simulators, and most of these are for the military. During the war, Delham simulators were used to train AT-6 and T-6 pilots. The device is just as effective in simulating at night as it is for night operations.

►Trouble-Maker—At the "engine" station at PAA's simulator, an instructor can create trouble by producing more than 40 mechanical malfunctions, including trouble with fuel flow, fuel pressure, fuel quantity, difficulties with oil pressure, carburetor icing, faulty



## Indian Overseas Folds

(McGraw-Hill World News)

Bombay—India's third airline collapse of the year is in the making, with the government supervisor of Indian Overseas Airlines (IOA) Bombay-Nagpur-Calcutta service had been operated on a daily basis, but lack of funds to pay for servicing of its Dakota and related frequency to less than half of schedule IOA's fleet consists of 15 Dakotas and one Skyvan, the latter bought for an untested British Australia service that never got underway. Latest

flight statistics, for May, put up the last a poor picture. Last month its planes flew 166,000 km but earned only 754 passengers and \$1,319 in freight.

## ATA Sees Upswing

An Transport Association predicts that the 16 scheduled domestic airlines will fly 6.5 billion passengers in the year—nearly 500 million more than in 1947, the previous record year. The estimate is based on a 15 percent passenger traffic increase in fiscal 1949 over the year 1948 period. It takes

into account a possible decline in traffic during the second half of 1949 but assumes that passenger business will continue substantially above the last six months of 1948.

## Pan American Forms Safety Group

Safety for flight crews is the objective of the Joint Flight Safety Committee, formed by pilots, engineers, radio operators, flight attendants and management of Pan American Airways Atlantic division.

Composed of five representatives from each of the safety committees of these groups, the organization intends to conduct research on subjects which will suggest safety procedures, equipment and policy for Atlantic flying.

First project of the committee is an attempt to clarify flight deck procedures. Underlining with full message event action, the group hopes to define and test action which should be taken in case of fire in various sections of the aircraft.

The committee is active only in the Atlantic division of Pan American. But the organization hopes that other divisions of the carrier, and other carriers, will form similar organizations.

Chairman of the Joint Flight Safety Committee is Peter De Remis, FAA instructor in emergency procedures. The committee was formed only in July and meets the last Monday of each month at closed sessions at LaGuardia Field in company furnished space.

## Cut Red Tape

Ways to cut travel and tape have been discussed by Pan American Airways' Systems Facilitation Committee, composed of representatives of the carrier's major divisions and foreign affiliates.

Harley H. Colburn, chairman of the committee, cited such countries as Argentina, which requires 21 visas at sea fees; facing the complete money of all passengers. "An airline is subject to 250 fine each time as retail is sold," Colburn reported.

The simplified documents covering international flights have been revised recently by International Civil Aviation Organization and adopted by 17 of the 21 American signatories, according to Colburn, but there are still a few holdouts.

As a tangible move to cut red tape, Board has voted as vital as necessary for travelers planning to visit in order to simplify country, provided they stay does not exceed eight days. Regulations apply to commercial and pleasure visitors from the western hemisphere and European countries.

## Hilred Reports On IATA Traffic

Financial problems of the world's airlines will tend to disappear since the carrier top the main transportation market, according to Sir Willem Hilred, director general of the International Air Transport Association.

Speaking at IATA's 30th anniversary meeting at The Hague, Holland, Hilred said that for the moment the main danger remains satisfactorily put out of reach.

This, he explained, is because of the carrier striving to keep on reducing fares in a period when costs continued to rise.

► **Month-Watering Market**—The IATA leaders described the United States as a particularly rich potential market of new foreign travel with its 20 million families in the \$3000 to \$5000 a year income bracket. "There is no reason why the amount of U. S. dollars put into international circulation through foreign travel could not be increased from the present level of \$600 million a year to \$2 billion. It is a market that could be made all year round water." IATA said 280 jet flights of 70 IATA member airlines.

During 1948, IATA airlines flew 12 billion passenger miles, 185 million tons miles of cargo and 750 million ton miles of mail. Hilred declared in his annual report. The figure represents a substantial increase over 1947, with the most impressive gain registered in cargo.

► **Schedule Needed**—But the IATA director pointed out that with still missing costs, international carriers must continue to rely on government subsidies to fly at all. "We could no doubt tap the mass market available if we could finance our operations, but at such time government agencies (as the U. S. Civil Aeronautics Board) have indicated we ought not go further in reducing rates at the present time."

Hilred's report urged more flexibility in rate accommodations of the IATA flight conferences to promote air action and open travel and to take advantage of special rates. It asked utmost advice support for efforts to reduce and simplify red tape in customs, immigration and other border formalities.

Other recommendations were: greater attention by the carriers and their governments to problems presented by new jet aircraft; a concerted effort to close the gap between the growing volume of international air transport and the development of airports, landing and navigation aids, multilateral agreements as all transport rights between airlines, instead of the present limited system of bilateral pacts, and a delegation of

## Safeguarding Aircraft Electrical Systems



Photo shows how circuit protection is provided for the R-28 by the installation of Burndy Type 115 Limiters.

As a safeguard to flight, today's modern aircraft electrical system is Burndy-Limiter protected. These vital accurate "fuses" carry temporary overloads but close promptly under short circuits. They are particularly recommended for systems which use multiple conductors per phase bus, where Limiter protected, a fault on a single wire is cleared without major disruption of current in the rest of the circuit.

The close co-ordination of these highly accurate Limiters, unlike other thermal devices is broke affected by the variation of ambient temperatures, thus they provide greater protection with the least weight and space.

Limiters and mountings are offered for 30-volt and 135-volt DC, and 120/208-volt, 480 cycle AC systems in various ampere ratings. Burndy Limiters meet the requirements of USAF Spec. Nos. 32932-A, 32906-B.

Complete engineering service is offered. For particulars, write for Bulletin 471.

Connect with  
**BURNDY**

New York 24, N. Y.

WESTERN MARKS: Toronto 11, Canada • CANADA: Canadian Line Machines, Ltd., Toronto 18

where precision counts . . .

# BH

**OIL PUMP FOR PRATT & WHITNEY R-2800 ENGINE**

- Gas Turbine Components
- Injector Pumps
- Fuelizer Cells
- Combs
- Collector Rings
- Regain Motors
- Aluminum Tools
- Diesel Motor Fuelizers
- Diesel Motor Overhaul

UP TO KNOW YOUR REQUIREMENTS

Supplies to manufacturers only

Patented under U. S. P.

Patented under U. S. P.

**H. H. AIRCRAFT COMPANY, Inc.**  
1000 Broadway, New York 10, N. Y.





## IMMEDIATE SALE!

## DC-4

(Type C-48 DC)

Interior Specially Arranged and  
Universally Furnished for Extreme Use

ACCOMMODATES 44 PASSENGERS

Comfortable bays and seats  
with lighting, heaters and telephonesExtremely finished and first class, always used  
for long-range and domestic airline serviceFour 3-place bays, telephones, special  
lighting, two lavatories, stainless steel buffetHas recent 1000-hour inspection.  
To be delivered with certificate.  
P&W R-2000-7 Engines  
Price on Application

THE BABB COMPANY, INC.

444 Madison Avenue, New York

C.A.A.-APPROVED OVERHAULS  
Engines N.T.S.O. P & W N.T.S.O. Engines  
For Sale

## Without Exchange—Exchange Price

R-1830-82 . \$2,500-\$3,200 exchange  
R-1840-AN-1 . 2,250 . 1,750 exchange  
R-985-AN-1 or 2 1,450 . 1,450 exchange  
R-985-140 . 2,500 . 1,850 exchange  
R-2000-81 or 2 2,150 . 1,850 exchange  
R-1820-45-92 . 1,850 Cash, Low Curb.

## Basic Overhaul Prices

R-1830's . . . . . \$1,200.00  
R-1840's . . . . . 700.00  
R-985's . . . . . 600.00  
R-2000's . . . . . 1,400.00  
The above prices are all Plus  
Parts

## DON'T TAKE CHANCES ON SURPLUS ENGINES

All work and engine sales carry our 100 hr. warranty

Shipment on Quantity Purchase of Five or More Engines

The management and Ken Mac of A. C. E. S. are all former Pratt & Whitney Aircraft  
Supervisors and have the benefit of over 75 years accumulated experience. We have  
complete facilities for overhaul, block fast, installation and handle all approvals  
concerning to the engine. We have been engaged in the Overhaul & Sale of engines  
here for the past four years.

CAA Approved Sta. #3404

AIR CARRIER ENGINE SERVICE, INC.

P. O. Box 1388

Fullerton #251 & 252 International Airport Miami, Florida  
Cable "ACENSRV"AIRCRAFT & ELECTRONIC  
EQUIPMENTAs a leading supplier we offer a  
complete line of

## BRAND NEW INSTRUMENTS

- \* FLIGHT & NAVIGATION INSTRUMENTS
- \* ENGINE INSTRUMENTS
- \* AUTOMATIC PILOTS
- \* SPEED LOGS
- \* ALTITUDE
- \* PRESSURE ALTITUDE
- \* RPT. GENERATORS
- \* STROBES
- \* MASTER FIELD SYSTEMS
- \* SPOOLS
- \* A.C. MOTORS
- \* D.C. MOTORS
- \* MAGNETO UNITS
- \* IGN. & AMP. SYSTEMS
- \* FREQUENCY METERS
- \* ROTARY SWITCHES

Write for complete listings  
All instruments ship by Air Parcel  
S. A. 60-500  
S. S. Export Carton 2140

## INSTRUMENT ASSOCIATES

37 E. Bayview Ave. Great Neck, N. Y.  
Tel. (Mead) 7-1147

## ENGINE SALE

12 Wright R-3350-23A ENGINES

1000 hrs. overhaul, complete in service  
condition. \$12000 ea.Lowe Engineering & Supply Co.  
New Orleans, La. New Orleans, La.

## Commercial Overhaul

R-1830-92 \*2125\* EXCHANGE  
R-985 ANI \*1300 EXCHANGE  
R-985 140 \*1350 EXCHANGE

\*Lower rates on contract Basis

Warranty extends full period CAA operating allowance between  
overhaul \* 18 Years authorized Pratt & Whitney parts  
and overhaul station \* CAA approved No. 88 since 1928

Write your nearest PAC Division \* Prices F.O.B. Burbank or Linden

PACIFIC AIRMOTIVE CORP. \* Burbank, Calif. \* Linden, N. J.

Other Divisions at Oakland, Seattle, Anchorage and Kansas City, Kansas

FOR SALE  
Two DC-3's PAN Engines 24 Pass.  
51000

Excellent Condition. Overhaul from 0-47

Price \$75,000.00

NATIONWIDE AIRLINES, INC.  
Miami Beach Airport, Miami & ChicagoCARLE ASSEMBLIES  
for DOUGLAS AIRCRAFT  
models such as 440, 440-1, 440-2, 440-3, 440-4, 440-5, 440-6, 440-7, 440-8, 440-9, 440-10, 440-11, 440-12, 440-13, 440-14, 440-15, 440-16, 440-17, 440-18, 440-19, 440-20, 440-21, 440-22, 440-23, 440-24, 440-25, 440-26, 440-27, 440-28, 440-29, 440-30, 440-31, 440-32, 440-33, 440-34, 440-35, 440-36, 440-37, 440-38, 440-39, 440-40, 440-41, 440-42, 440-43, 440-44, 440-45, 440-46, 440-47, 440-48, 440-49, 440-50, 440-51, 440-52, 440-53, 440-54, 440-55, 440-56, 440-57, 440-58, 440-59, 440-60, 440-61, 440-62, 440-63, 440-64, 440-65, 440-66, 440-67, 440-68, 440-69, 440-70, 440-71, 440-72, 440-73, 440-74, 440-75, 440-76, 440-77, 440-78, 440-79, 440-80, 440-81, 440-82, 440-83, 440-84, 440-85, 440-86, 440-87, 440-88, 440-89, 440-90, 440-91, 440-92, 440-93, 440-94, 440-95, 440-96, 440-97, 440-98, 440-99, 440-100

A. J. McGee, CO.

240 7th Ave. S. W. (1), N. Y.

Phone (212) 600-0000





## EDITORIAL

### CAB's Air Coach Decision—An Analysis

Airline passengers of skycoach and other types of turboprop aircraft have won a significant victory among Civil Aeronautics Board policy-makers.

This month, for the first time, CAB conceded that four-craft-wide turboprop service between intermediate coastal cities can produce sufficient new traffic to make coach service profitable on congested routes. As a result, it sanctioned the continuance through next June of non-stop, tourist-type operations conducted with high-capacity equipment during off-peak traffic periods (ANIMATOR, Week, Sept. 13).

But the victory for skycoach was by no means a complete one. CAB also decided that four-craft-wide service with low-capacity equipment such as 24- or 34-passenger DC-3s probably cannot be conducted at a profit unless almost untamable load factors (approach of 85 percent) are maintained over an extended period.

"There is little indication," the Board asserted, "that the airline industry is in a position to enter the air coach less than on a local and rudimentary basis. Despite the favorable trend of airline costs and the several options in savings, the present economic position of the certificate carriers appears to demand considerations of saving fuel loads for the great bulk of passenger air travel."

Successful and economic through operations apparently require these conditions, according to CAB.

Heavy traffic flow must characterize routes.

High-density equipment (with more than the average number of seats) must be used.

Service must be scheduled to minimize diversion of traffic from regular flights.

All non-essential services to the passenger, such as in-flight meals, extra stewardesses, full stewardess procedures, etc., must be eliminated.

The Board decided that the DC-3 coach operations of TWA between Kansas City and Los Angeles and Continental Air Lines' four-craft-wide DC-1 can between Kansas City and Denver did not meet the standards specified. As a result, Continental's tariff will not be extended beyond Sept. 30. TWA was permitted to continue its DC-3 coach service between Kansas City and Los Angeles through Dec. 31 to give the company time to explore the possibility of using high density four-engine equipment in the operation.

But the coach service CAB will permit to continue to June 30, together with the new operations it will approve, overhauled operation of the TWA and Continental flights.

Capital Airlines, first certificated domestic carrier to offer tourist-type service, will continue to make its "Nighthawk" flights from Washington and New York to Chicago and Minneapolis-St. Paul with 39-passenger DC-4s. TWA's New York-Chicago coach flights with 35-passenger Boeing Stratliners and Northwest Airlines' New York-Salt Lake coach run with 35-passenger DC-4s were also approved for a nine-month extension.

In addition, CAB said it was prepared to permit Western Air Lines to operate a new 40-passenger DC-4 coach service between San Diego and Seattle (to depart twice) or between Los Angeles and Seattle (to depart twice) between 10 p.m. and 1 a.m. Suspension of Northwest

Airlines' proposed Chicago-Portland, Ore., low-cost flights with 35-seat combination passenger/cargo DC-4s will be lifted.

The Board rejected Capital Airlines' request to extend Nighthawk flights to the New York-Albany-New Orleans run with 39-passenger DC-4s and Northwest Airlines' proposed New York-Miami coach service with 40-passenger DC-4s. Capital successfully proposed to settle its tariff to provide for use of high-density 39-passenger DC-4s on the proposed southwest extension of Nighthawk service.

Capital Airlines officials told ANIMATOR, Week, they were pleased by CAB's policy statement on skycoach. They noted that the principles for successful coach-type operations set down by CAB are almost identical with those adopted by Capital when it began Nighthawk flights last November. The nine-month extension of coach traffic, Capital pointed out, is by far the largest affirmation of this type to be made by CAB.

Coach traffic itself in the future will be given careful consideration by CAB if they meet the four conditions outlined. But the federal agency emphasized that it expects such coach lines to be accompanied and supported by detailed cost-conscious justification both from a traffic and cost standpoint.

"The burden of proof for establishment of new coach services is clearly on the carrier," CAB said. "We do not propose to allow the indiscriminate extension of such lines, so we intend to grant a general endorsement of the existing passenger fare level."

Turning to other potential traffic, the Board declared that the last-of-the-week family fare plan (providing 50 percent reduction on Mondays, Tuesdays and Wednesdays) appears to have been successful in building up traffic during periods when business is normally light. It noted, however, that the degree to which the plan generates new business—as opposed to diverting traffic from a peak to an off-peak period of the week—is not entirely clear.

Although available data for the summer season are not yet available, CAB denied the family fare has generated enough new traffic to warrant continuation of the plan through next June. The Board said it was not ready to approve Continental Air Lines' proposal for a family plan which would be applicable to every day of the week.

Western Air Lines and Inland Air Lines will be permitted to continue their "no-meal tariff" for another nine months. Adopted by the two carriers last February, this measure provided for a 5 percent fare cut in lieu of in-flight meal service.

CAB said Western's continuing regarding concerns resulting from elimination of much service to have been borne out. The Board noted, however, that the effect of the no-meal tariff on Western/Inland business is still uncertain.

Since CAB regards the whole potential fare structure as an experiment, it will require each carrier with such tariffs to submit a monthly interest-free record of results. In this way CAB hopes to have a firm basis for determining which promotional fare device should be a permanent part of the domestic passenger fare structure.

# Bendix Products

FUEL METERING SYSTEMS and COMPLETE LANDING GEAR



**Creative Engineering that Helps American Aviation Lead the World**

Building complete landing gear for the world's largest land plane, or developing fuel metering systems for the latest jet is all part of the job for Bendix Products Division of the great Bendix Aviation Corporation. For here under one roof are the men and machines that have over the years furnished the landing gear and fuel systems which help American aviation lead the world.

Engine builders and airplane manufacturers are agreed to let this matchless combination of engineering experience and manufacturing facilities help solve their problems.

## BENDIX PRODUCTS

DIVISION of

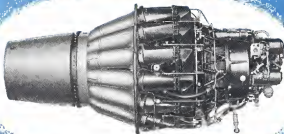


SOUTH BEND 20, INDIANA

Export Sales, Bendix Aviation Corp., 12-13, 15-16, 17-18, 19-20, 21-22, 23-24, 25-26, 27-28, 29-30, 31-32, 33-34, 35-36, 37-38, 39-40, 41-42, 43-44, 45-46, 47-48, 49-50, 51-52, 53-54, 55-56, 57-58, 59-60, 61-62, 63-64, 65-66, 67-68, 69-70, 71-72, 73-74, 75-76, 77-78, 79-80, 81-82, 83-84, 85-86, 87-88, 89-90, 91-92, 93-94, 95-96, 97-98, 99-100, 101-102, 103-104, 105-106, 107-108, 109-110, 111-112, 113-114, 115-116, 117-118, 119-120, 121-122, 123-124, 125-126, 127-128, 129-130, 131-132, 133-134, 135-136, 137-138, 139-140, 141-142, 143-144, 145-146, 147-148, 149-150, 151-152, 153-154, 155-156, 157-158, 159-160, 161-162, 163-164, 165-166, 167-168, 169-170, 171-172, 173-174, 175-176, 177-178, 179-180, 181-182, 183-184, 185-186, 187-188, 189-190, 191-192, 193-194, 195-196, 197-198, 199-200, 201-202, 203-204, 205-206, 207-208, 209-210, 211-212, 213-214, 215-216, 217-218, 219-220, 221-222, 223-224, 225-226, 227-228, 229-230, 231-232, 233-234, 235-236, 237-238, 239-240, 241-242, 243-244, 245-246, 247-248, 249-250, 251-252, 253-254, 255-256, 257-258, 259-260, 261-262, 263-264, 265-266, 267-268, 269-270, 271-272, 273-274, 275-276, 277-278, 279-280, 281-282, 283-284, 285-286, 287-288, 289-290, 291-292, 293-294, 295-296, 297-298, 299-300, 301-302, 303-304, 305-306, 307-308, 309-310, 311-312, 313-314, 315-316, 317-318, 319-320, 321-322, 323-324, 325-326, 327-328, 329-330, 331-332, 333-334, 335-336, 337-338, 339-340, 341-342, 343-344, 345-346, 347-348, 349-350, 351-352, 353-354, 355-356, 357-358, 359-360, 361-362, 363-364, 365-366, 367-368, 369-370, 371-372, 373-374, 375-376, 377-378, 379-380, 381-382, 383-384, 385-386, 387-388, 389-390, 391-392, 393-394, 395-396, 397-398, 399-400, 401-402, 403-404, 405-406, 407-408, 409-410, 411-412, 413-414, 415-416, 417-418, 419-420, 421-422, 423-424, 425-426, 427-428, 429-430, 431-432, 433-434, 435-436, 437-438, 439-440, 441-442, 443-444, 445-446, 447-448, 449-450, 451-452, 453-454, 455-456, 457-458, 459-460, 461-462, 463-464, 465-466, 467-468, 469-470, 471-472, 473-474, 475-476, 477-478, 479-480, 481-482, 483-484, 485-486, 487-488, 489-490, 491-492, 493-494, 495-496, 497-498, 499-500, 501-502, 503-504, 505-506, 507-508, 509-510, 511-512, 513-514, 515-516, 517-518, 519-520, 521-522, 523-524, 525-526, 527-528, 529-530, 531-532, 533-534, 535-536, 537-538, 539-540, 541-542, 543-544, 545-546, 547-548, 549-550, 551-552, 553-554, 555-556, 557-558, 559-560, 561-562, 563-564, 565-566, 567-568, 569-570, 571-572, 573-574, 575-576, 577-578, 579-580, 581-582, 583-584, 585-586, 587-588, 589-590, 591-592, 593-594, 595-596, 597-598, 599-600, 601-602, 603-604, 605-606, 607-608, 609-610, 611-612, 613-614, 615-616, 617-618, 619-620, 621-622, 623-624, 625-626, 627-628, 629-630, 631-632, 633-634, 635-636, 637-638, 639-640, 641-642, 643-644, 645-646, 647-648, 649-650, 651-652, 653-654, 655-656, 657-658, 659-660, 661-662, 663-664, 665-666, 667-668, 669-670, 671-672, 673-674, 675-676, 677-678, 679-680, 681-682, 683-684, 685-686, 687-688, 689-690, 691-692, 693-694, 695-696, 697-698, 699-700, 701-702, 703-704, 705-706, 707-708, 709-710, 711-712, 713-714, 715-716, 717-718, 719-720, 721-722, 723-724, 725-726, 727-728, 729-730, 731-732, 733-734, 735-736, 737-738, 739-740, 741-742, 743-744, 745-746, 747-748, 749-750, 751-752, 753-754, 755-756, 757-758, 759-760, 761-762, 763-764, 765-766, 767-768, 769-770, 771-772, 773-774, 775-776, 777-778, 779-780, 781-782, 783-784, 785-786, 787-788, 789-790, 791-792, 793-794, 795-796, 797-798, 799-800, 801-802, 803-804, 805-806, 807-808, 809-810, 811-812, 813-814, 815-816, 817-818, 819-820, 821-822, 823-824, 825-826, 827-828, 829-830, 831-832, 833-834, 835-836, 837-838, 839-840, 841-842, 843-844, 845-846, 847-848, 849-850, 851-852, 853-854, 855-856, 857-858, 859-860, 861-862, 863-864, 865-866, 867-868, 869-870, 871-872, 873-874, 875-876, 877-878, 879-880, 881-882, 883-884, 885-886, 887-888, 889-890, 891-892, 893-894, 895-896, 897-898, 899-900, 901-902, 903-904, 905-906, 907-908, 909-910, 911-912, 913-914, 915-916, 917-918, 919-920, 921-922, 923-924, 925-926, 927-928, 929-930, 931-932, 933-934, 935-936, 937-938, 939-940, 941-942, 943-944, 945-946, 947-948, 949-950, 951-952, 953-954, 955-956, 957-958, 959-960, 961-962, 963-964, 965-966, 967-968, 969-970, 971-972, 973-974, 975-976, 977-978, 979-980, 981-982, 983-984, 985-986, 987-988, 989-990, 991-992, 993-994, 995-996, 997-998, 999-1000, 1001-1002, 1003-1004, 1005-1006, 1007-1008, 1009-1010, 1011-1012, 1013-1014, 1015-1016, 1017-1018, 1019-1020, 1021-1022, 1023-1024, 1025-1026, 1027-1028, 1029-1030, 1031-1032, 1033-1034, 1035-1036, 1037-1038, 1039-1040, 1041-1042, 1043-1044, 1045-1046, 1047-1048, 1049-1050, 1051-1052, 1053-1054, 1055-1056, 1057-1058, 1059-1060, 1061-1062, 1063-1064, 1065-1066, 1067-1068, 1069-1070, 1071-1072, 1073-1074, 1075-1076, 1077-1078, 1079-1080, 1081-1082, 1083-1084, 1085-1086, 1087-1088, 1089-1090, 1091-1092, 1093-1094, 1095-1096, 1097-1098, 1099-1100, 1101-1102, 1103-1104, 1105-1106, 1107-1108, 1109-1110, 1111-1112, 1113-1114, 1115-1116, 1117-1118, 1119-1120, 1121-1122, 1123-1124, 1125-1126, 1127-1128, 1129-1130, 1131-1132, 1133-1134, 1135-1136, 1137-1138, 1139-1140, 1141-1142, 1143-1144, 1145-1146, 1147-1148, 1149-1150, 1151-1152, 1153-1154, 1155-1156, 1157-1158, 1159-1160, 1161-1162, 1163-1164, 1165-1166, 1167-1168, 1169-1170, 1171-1172, 1173-1174, 1175-1176, 1177-1178, 1179-1180, 1181-1182, 1183-1184, 1185-1186, 1187-1188, 1189-1190, 1191-1192, 1193-1194, 1195-1196, 1197-1198, 1199-1200, 1201-1202, 1203-1204, 1205-1206, 1207-1208, 1209-1210, 1211-1212, 1213-1214, 1215-1216, 1217-1218, 1219-1220, 1221-1222, 1223-1224, 1225-1226, 1227-1228, 1229-1230, 1231-1232, 1233-1234, 1235-1236, 1237-1238, 1239-1240, 1241-1242, 1243-1244, 1245-1246, 1247-1248, 1249-1250, 1251-1252, 1253-1254, 1255-1256, 1257-1258, 1259-1260, 1261-1262, 1263-1264, 1265-1266, 1267-1268, 1269-1270, 1271-1272, 1273-1274, 1275-1276, 1277-1278, 1279-1280, 1281-1282, 1283-1284, 1285-1286, 1287-1288, 1289-1290, 1291-1292, 1293-1294, 1295-1296, 1297-1298, 1299-1300, 1301-1302, 1303-1304, 1305-1306, 1307-1308, 1309-1310, 1311-1312, 1313-1314, 1315-1316, 1317-1318, 1319-1320, 1321-1322, 1323-1324, 1325-1326, 1327-1328, 1329-1330, 1331-1332, 1333-1334, 1335-1336, 1337-1338, 1339-1340, 1341-1342, 1343-1344, 1345-1346, 1347-1348, 1349-1350, 1351-1352, 1353-1354, 1355-1356, 1357-1358, 1359-1360, 1361-1362, 1363-1364, 1365-1366, 1367-1368, 1369-1370, 1371-1372, 1373-1374, 1375-1376, 1377-1378, 1379-1380, 1381-1382, 1383-1384, 1385-1386, 1387-1388, 1389-1390, 1391-1392, 1393-1394, 1395-1396, 1397-1398, 1399-1400, 1401-1402, 1403-1404, 1405-1406, 1407-1408, 1409-1410, 1411-1412, 1413-1414, 1415-1416, 1417-1418, 1419-1420, 1421-1422, 1423-1424, 1425-1426, 1427-1428, 1429-1430, 1431-1432, 1433-1434, 1435-1436, 1437-1438, 1439-1440, 1441-1442, 1443-1444, 1445-1446, 1447-1448, 1449-1450, 1451-1452, 1453-1454, 1455-1456, 1457-1458, 1459-1460, 1461-1462, 1463-1464, 1465-1466, 1467-1468, 1469-1470, 1471-1472, 1473-1474, 1475-1476, 1477-1478, 1479-1480, 1481-1482, 1483-1484, 1485-1486, 1487-1488, 1489-1490, 1491-1492, 1493-1494, 1495-1496, 1497-1498, 1499-1500, 1501-1502, 1503-1504, 1505-1506, 1507-1508, 1509-1510, 1511-1512, 1513-1514, 1515-1516, 1517-1518, 1519-1520, 1521-1522, 1523-1524, 1525-1526, 1527-1528, 1529-1530, 1531-1532, 1533-1534, 1535-1536, 1537-1538, 1539-1540, 1541-1542, 1543-1544, 1545-1546, 1547-1548, 1549-1550, 1551-1552, 1553-1554, 1555-1556, 1557-1558, 1559-1560, 1561-1562, 1563-1564, 1565-1566, 1567-1568, 1569-1570, 1571-1572, 1573-1574, 1575-1576, 1577-1578, 1579-1580, 1581-1582, 1583-1584, 1585-1586, 1587-1588, 1589-1590, 1591-1592, 1593-1594, 1595-1596, 1597-1598, 1599-1600, 1601-1602, 1603-1604, 1605-1606, 1607-1608, 1609-1610, 1611-1612, 1613-1614, 1615-1616, 1617-1618, 1619-1620, 1621-1622, 1623-1624, 1625-1626, 1627-1628, 1629-1630, 1631-1632, 1633-1634, 1635-1636, 1637-1638, 1639-1640, 1641-1642, 1643-1644, 1645-1646, 1647-1648, 1649-1650, 1651-1652, 1653-1654, 1655-1656, 1657-1658, 1659-1660, 1661-1662, 1663-1664, 1665-1666, 1667-1668, 1669-1670, 1671-1672, 1673-1674, 1675-1676, 1677-1678, 1679-1680, 1681-1682, 1683-1684, 1685-1686, 1687-1688, 1689-1690, 1691-1692, 1693-1694, 1695-1696, 1697-1698, 1699-1700, 1701-1702, 1703-1704, 1705-1706, 1707-1708, 1709-1710, 1711-1712, 1713-1714, 1715-1716, 1717-1718, 1719-1720, 1721-1722, 1723-1724, 1725-1726, 1727-1728, 1729-1730, 1731-1732, 1733-1734, 1735-1736, 1737-1738, 1739-1740, 1741-1742, 1743-1744, 1745-1746, 1747-1748, 1749-1750, 1751-1752, 1753-1754, 1755-1756, 1757-1758, 1759-1760, 1761-1762, 1763-1764, 1765-1766, 1767-1768, 1769-1770, 1771-1772, 1773-1774, 1775-1776, 1777-1778, 1779-1780, 1781-1782, 1783-1784, 1785-1786, 1787-1788, 1789-1790, 1791-1792, 1793-1794, 1795-1796, 1797-1798, 1799-1800, 1801-1802, 1803-1804, 1805-1806, 1807-1808, 1809-1810, 1811-1812, 1813-1814, 1815-1816, 1817-1818, 1819-1820, 1821-1822, 1823-1824, 1825-1826, 1827-1828, 1829-1830, 1831-1832, 1833-1834, 1835-1836, 1837-1838, 1839-1840, 1841-1842, 1843-1844, 1845-1846, 1847-1848, 1849-1850, 1851-1852, 1853-1854, 1855-1856, 1857-1858, 1859-1860, 1861-1862, 1863-1864, 1865-1866, 1867-1868, 1869-1870, 1871-1872, 1873-1874, 1875-1876, 1877-1878, 1879-1880, 1881-1882, 1883-1884, 1885-1886, 1887-1888, 1889-1890, 1891-1892, 1893-1894, 1895-1896, 1897-1898, 1899-1900, 1901-1902, 1903-1904, 1905-1906, 1907-1908, 1909-1910, 1911-1912, 1913-1914, 1915-1916, 1917-1918, 1919-1920, 1921-1922, 1923-1924, 1925-1926, 1927-1928, 1929-1930, 1931-1932, 1933-1934, 1935-1936, 1937-1938, 1

**MORE THRUST PER DOLLAR**

*Allison aircraft turbine engines deliver more thrust per dollar  
of engine cost than any other jet engines in the world.*



*J 33-A-23 Turbo-Jet*

**Allison jet-powered airplanes:**

*Lockheed F-80 and TF-80 Shooting Star  
Republic F-84 Thunderjet  
Grumman F9F-3 Panther  
Northrop F-89A Scorpion  
North American FJ-1 Fury  
Lockheed F-94  
Consolidated XP5Y  
Northrop RB-35B Flying Wing  
Martin P4M-1 Mercator  
North American AJ-1*

*Allison*

Builder of axial and centrifugal flow turbine engines



DIVISION OF

Indianapolis, Indiana